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# AVIATION

*The Oldest American Aeronautical Magazine*



## *Pratt & Whitney* **TO POWER TRANS-CANADA AIR LINES**

1938 will mark the inauguration of another great trans-continental air line. Paralleling the famous U. S. routes, Trans-Canada Air Lines will offer our northern neighbors a comparable service in swift transportation. The fleet will consist of Lockheed planes . . . all powered by Pratt & Whitney engines.

**PRATT & WHITNEY AIRCRAFT**  
EAST HARTFORD, CONNECTICUT

*Division of*  
**UNITED AIRCRAFT CORPORATION**



No. 12 of a series of aluminum  
construction drawings, furnished  
at Alcoa's request to  
investigate "High Cycle Tools"  
Pittsburgh, Pa.

# PX

## December 1937

With traffic increased, with costs going up, the drive for efficiency and sound economy by both operating and manufacturing sides of the industry has been even stronger.

As a result, these trends in the use of Alcoa Aluminum have been discernible:

Increased use of bare Alclad sheets, to eliminate the expense and weight of separate protective coating on exposed parts.

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We consider it a compliment by other members of the industry that these practices are now being accepted. These were developed and introduced several years ago, and adoption today, after thorough scrutiny, lends them sound. Aluminum Company of America will continue its program looking to improved methods and materials which will make further contributions to maintenance and production economy for the industry.



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AVIATION  
Division 302

# Are High Cycle Tools accepted by Industry - - ?



## Certainly! . . . . HIGH CYCLE IS THE MODERN METHOD OF POWERING PORTABLE TOOLS

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# The Birdmen's Perch

Now that we agree to think of E. E.'s letter, what DOES happen when a group makes accidental moves out of instinct "on the wing"? If any of you shop-hoppers have ever found this reaction at just what happens when a bird flies into a group, wouldn't it be a good idea?

MAURIE AL. WILSON (aka "Tarned Wing-Tip") (aka "Auntie Bird", Gold Avenue, Pittsburgh, Pa.)



## CORRESPONDENCE WANTED FROM KANSASLAND

I am anxious to strike up a correspondence with some one, under the age of twenty, less an aviation, residing in U.S.A. who has positive no less. I am interested in all free hand activities concerning airplanes.

Frank Higgins, 127 Hawthorne St., Council, S.E. 2, Oklahoma City.

## MAN VS. BIRD

"It's a little tricky about having my memory checked—and I've gotten the bird over!"

I can't tell exactly how the first one happened, because I was flying blind in the zone, but I felt the sensation of a jerk and a sensation of the moon—then everything kept on smoothly.

When I got down, I found eagle feathers on my desk in open rooms, etc. And again, if there was any of collecting eagle feathers now into up in the air, outside of clipping on eagle with my pen, I don't know about it.

I got the second "bird" because the boys



checked it should have been more recent instead of the eagle. How about putting in a word for me?

—E. E. Brinkley, N.Y.

## ARE NAVIGATORS SMART?

"Tarned Wing-Tip" is a smart! But the following text of ignorance pointed out a suitable remedy. Out of five flying girls, one was smart at it all, three solved it with pencil and paper, while one navigator got solved it in his head.

"T W T" thought reader might like a sketch of the incident. Here goes: At breakfast, I found directly on my hand, found a tiny yellow spot on the ground which caught my attention. I decided to take another look, and, and, and, and



the wind, caught up in the balloon again in order for the eagle. He had an enormous white tail and white wings!

"T W T" was glad to acknowledge the queen as to the correct answer.

## WINGS ON AUTOS

To get that simple performance from your auto, just try Gulf Oil-War Goodies Gulf-Max cars also have what it takes to put wings on your car!

## REFINED BODY DESI

On the Gulf Oil-War Goodies Gulf-Max cars, to be sure, the complete Gulf-Max cars also have what it takes to put wings on your car!

## THIS MONTH'S WHOPPER

"I, Seagull, am your Rooster's dancer who held world record for no jump, to look and to spin, in a moment's time. I had much trouble with my wings, but Gulf Oil-War Goodies Gulf-Max cars also have what it takes to put wings on your car!"

Oh, you, who is dancer partner, no like so much. I think the reader before dance, but, because I, Seagull, make Seagull to spin like a top, to look back, like a top. Sometimes I get lost in a pool when he spins the look of Seagull in the by purpose.

AVIATION December, 1937

# New Wings over CANADA



## FASTEST Coast to Coast Schedule in North America

Seizing an advantage with the most advanced type transport yet developed, Trans-Canada Airlines is pioneering a transcontinental route with the fastest flight schedules in North America. New Lockheed "14" transports have been selected. This equipment is 35 miles per hour faster than planes used by U. S. airlines flying from coast to coast.

With luxurious accommodations for 18 passengers and a crew of three, and the greatest cargo capacity ever built into a plane of its size, the "14", with a top speed of 250 m. p. h., is the fastest transport built. LOCKHEED AIRCRAFT CORP., Burbank, Calif. New York, 614 Chrysler Building. Chicago, 2353 Field Building. Dallas, Love Field.

Gulf Oil Corporation and Gulf Refining Company... makers of



Lockheed 14, 14A, 14B, 14C, 14D, 14E, 14F, 14G, 14H, 14I, 14J, 14K, 14L, 14M, 14N, 14O, 14P, 14Q, 14R, 14S, 14T, 14U, 14V, 14W, 14X, 14Y, 14Z, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52, 53, 54, 55, 56, 57, 58, 59, 60, 61, 62, 63, 64, 65, 66, 67, 68, 69, 70, 71, 72, 73, 74, 75, 76, 77, 78, 79, 80, 81, 82, 83, 84, 85, 86, 87, 88, 89, 90, 91, 92, 93, 94, 95, 96, 97, 98, 99, 100.

14

Lockheed 14, 14A, 14B, 14C, 14D, 14E, 14F, 14G, 14H, 14I, 14J, 14K, 14L, 14M, 14N, 14O, 14P, 14Q, 14R, 14S, 14T, 14U, 14V, 14W, 14X, 14Y, 14Z, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52, 53, 54, 55, 56, 57, 58, 59, 60, 61, 62, 63, 64, 65, 66, 67, 68, 69, 70, 71, 72, 73, 74, 75, 76, 77, 78, 79, 80, 81, 82, 83, 84, 85, 86, 87, 88, 89, 90, 91, 92, 93, 94, 95, 96, 97, 98, 99, 100.

# Already the Greatest Air Show

More—than the **1936** **Air Show** is **admission-free** features of the 1938 International Air Show in Chicago make this the most successful trade and popular aviation show ever held. It is the largest show in actual physical size, in number and variety of aircraft shown, in completeness of representation by aircraft and aircraft manufacturers, in educational features, in technical and technical features, and in opportunity for productive sales effort. **Plan now to take part in the one big show of the year—the show in the amphitheatre possible only at the 1938 International Air Show.**

**CLASS "A" SHOW** The 1938 International Air Show at Chicago is the first Class "A" aviation show ever held. Sponsored by Associated Chambers of Commerce of America, Inc.

**COMPLETE** For the first time, the exhibits and features of this show will present a complete picture of the aviation industry, including private, commercial and transport, and military aviation.

**MORE EXHIBITS** Present space requirements exceed past years by far more aircraft manufacturers than of any previous show. Engines, parts and accessory manufacturers, and service organizations will be most completely represented. Many commercial aviation exhibits will display their products in the show.

**MORE MODELS** None the spectacular display of giant military and commercial ships to the exhibits of the popular light planes, the 1938 Air Show will present every variety of present-day aircraft. The majority of manufacturers will exhibit at least two models. Many will show three or more different types.

**EDUCATION** An entire floor (over 80,000 sq. ft.) will be devoted to educational displays. Included will be exhibits of the Army, Navy, Coast Guard, Bureau of Air Commerce, and National Advisory Committee for Aeronautics.

**AIR TRANSPORT** The leading air transport operators are preparing comprehensive exhibits showing the latest scientific equipment for airway air transportation.

**SPECTACULAR AVIATION SHOW** The entire Arena of the International Amphitheatre will be devoted to a display of giant military and commercial airplanes of the most modern types, including in Douglas Supermarine Pioneers, Air King Bombers, and other types exhibited all day. A spectacular show will be presented on the Arena stage, with two performances each day.

*Make your plans NOW!*

There is still space available for a representative exhibit of your products, still time to arrange for participation in the 1938 International Air Show. Write or telegraph for detailed information on space available.

# in the History of the Industry!

## ADVISORY BOARD

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**JANUARY 28 • FEBRUARY 6, 1938**

UNDER YEARS, President. LAC & GIBBS, Inc., Executive Vice President. MULLIGAN & GIBBS, Inc., Vice President.

AVIATION  
December, 1937

AVIATION  
December, 1937





## 210 of These Attack Planes Now In U.S. Service

The G.H.Q. maneuvers in California last May again demonstrated the supremacy of Northrop Model A-17 Attackers. With their 750 h.p. Pratt & Whitney "Twin Wasp Jr." geared engines and three-blade Hamilton Standard Propellers, these 30 caliber machine guns and bomb loads of over 1000 lbs., these great air fighters proved their importance to national defense. An actual fleet of 110 has now been organized by 100 of the Model A-17A with fully retractable landing gear.

DOUGLAS AIRCRAFT CO., INC.  
Northrop Division  
EL SEGUNDO, CALIF.

It just means in fact that in the last two months the editorial staff of Aviation has traveled upwards of 15,000 miles here and there, practically all by air. Besides the visiting shoot conducted in the Cleveland Basin (Taylor Aircraft Taylor-Yount, Licensing: RCA, Flamingo, Lenoire, etc.) and our sacred Bermuda flight as reported in last month's *Flashes*, our West Coast Editor recently visited a transcontinental round trip which took him through the middle- and south-west areas, and your editor has just returned from a two week cruise into the northwest territory. Altogether, during the twelve months that end with the first of December, Aviation has covered perennally for its readers every corner of the United States and a large series of Europe in that time we have flown over most of the air routes in the country and abroad, and have visited many of the principal airports. And there are few more lacking plans where our projects are under way than we have noted. We're rather proud of the fact that Aviation just ran on a "pass out and return" policy. We're pretty much "Home Mission" when it comes to gathering material for the book.

McKerrow came out via TWA with a stop at Kansas City to see the maintenance base there. Westhead, he did a little flying and meeting (via TWA, Bristol, and a few ac-

count two feet), touching at Chicago, Kansas City, Wichita, Oklahoma City, Tulsa, Dallas, Ft. Worth, Amarillo, Albuquerque, and L. A. Every morning for two weeks our office staff helped with reports of Mac's findings in the field. Unfortunately we can't print them in full, but here are some notes:

✶ (Chicago) Boney Howard's plane going full blast with constant stops delivered in last year, including two just shipped off to Mexico Bay.

### In This Issue

Ed Allen is with us again with a discussion of some of the problems facing test pilots. ... St. Cloud, where with another of his series of lectures on aviation, ... How to shorten long and involved introductions by speaking periods in the course of two articles by Mr. Westhead, the first of which appears in this issue. The author is in the direct department of the Standard Company. ... Claude Ryan discusses some of the details of his deep practice on 30 production. ... Mac's new flying book and more interesting than the Westhead volume are his notes on this month's Flying Equipment. ... Just as we move about in the press is depicted in the photographs of the Miles Sherwin captured from our Danes Cove Squadron, TWA, Wichita, W.G. the help of a close team, we accepted in our first one day issue.

On page 32 of October issue the "World's Largest" Boeing plane of the Boeing plant should have been credited with 5000 BHP instead of 3000 pounds.

# Flashes

From the Skyways  
of the World

right to resume stages on production line. Plans call for 20,000 plane program for 1938. ... Aircraft show projects look good, a real setup with real looking. Show management: "hard hitting, intelligent, and an enormous job will take the greatest American show in history. ... a real industry opportunity. ... (Kansas City) Spent two interesting days with Houser Lindhouse, Ted LaFrance (Wichita), Bill Gray, the Rostrum, Ed. Porterville, Victor Ross, Schuchman, and TWA personnel. (Wichita) Had nice visit with Sam Rosenfeld and Ope Savage of Southern. They had a very interesting 20-page file on a very recent case, complete with ATC in test flight stages, some of which later.

See Lewis's production articles with Danes Westhead. ... A pleasant visit and a long talk with Earl Schaffner of Southern on production and report problems. ... (Book is bearing right along with three two motor in final stages and fourth and fifth under active construction. (TWA) Spent in active production of Brewster (see description, Aviation, November, 1937). ... Los Gatos at Spartan School sponsoring an interesting program of cross-country flying for his students. Presentation of five ships sent to Cleveland for the Race. ... (Dallas) Made impression by the constant and efficient efficiency of Brown's maintenance shop at Love Field. Ross Shuster, after his first reaction with Brown's "one of the sweetest







## Correlation!

How often, in the manufacture of all types of mechanisms, original designs are vastly improved by what might be called *Correlative Engineering*!

Component parts are altered to fit and to function better in relation, week to all others. Much of such valuable correlation can be observed perfectly beforehand through early consultation.

In the sincere desire to be of greater service, Bendix earnestly suggests to all aircraft engineering staffs that they take full and early advantage of Bendix' fund of specialized knowledge of Landing Gear design — to the end that all undercarriage units may be more perfectly correlated.

**BENDIX PRODUCTS CORPORATION**  
AIRPLANE WHEEL AND BRAKE DIVISION

(Subsidiary of Bendix Aviation Corporation) South Bend, Indiana

# BENDIX

AIRPLANE WHEELS • BRAKES • PILOT SEATS • PNEUMATIC SHOCK STRUTS



By  
**ROBERT  
OSBORN**

Presently it is just the working of nature's law of compensation or something, but we are always inclined to find that the smallest men carry the largest burdens, plane owners almost seem to weigh about 120 pounds, and the men who build the big dams and bridges and who build the big steel companies and other heavy industries generally are little men five feet high. On the other hand the scientists for most of laboratory, writers of poetry, and laboratory research workers seem to have been more nearly better men in case or lookalike.

All of which is brought to mind by the fact that Ed. Allen, who has in



us on his fellows to see over the edge of the coming and whose weight is doubled when he puts on a winter flying suit and parachute, is doing the right thing at most of the new large transport and bombers, including some of the world's largest airplanes.

Probably the heavier planes are

lighter loaded by men who need shoe horns to get themselves through the doors.

According to "THE AVIATIONIST" test flights have been completed in England on the two separate units of the Mayo Composite airplanes and tests of the combination are about to be started. In case you are not familiar with "May" development, the scheme is to have a hard heavily-loaded trim Atlantic airplane and a large lightly-loaded airplane attached together for the take off and after sufficient altitude and speed has been attained the smaller ship detaches itself and goes on its way.

If this invention is successful and is used on the regular trans-Atlantic service we can increase the pilot of the large launchers these companies at home to his wife about the moment of his entrance—"Take off, climb and land again all day long, and never get anywhere!"

Also, carrying this Mayo Composite airplane also a little further in the near future we can expect to see the twelve fifteen airplanes coming out on the wing of a large transport for delivery to distant customers, just as automobiles are delivered from the factory today.

WITH FEETAL AIRS BEING shown in the Middle Ages, the foreign-owned enterprises' home' destroyed and the design mental care being well-developed in China, all ending "by accident" and "because of mistakes identity" we are beginning to see what pilots mean when they ask for better vision in airplanes.

ALSO THIS SAME TIME we note that the rule-controlled gasoline fuel airplanes which Great Britain developed for use in anti-aircraft practice have been revised from that secret lot so that all nations will probably be using them. As the "vision" for anti-aircraft purposes might be as good as for being purely and honestly it would be well for commercial airplane designers to carefully avoid confusion with these other targets.

Now that the regular passenger service between Baltimore and New York is about to start operations we, as usual, hesitate to bring up the subject again, but, alas, now, about this job of ticket agent in Baltimore . . .

MANY AIRPLANE AND PASSENGER, looking being a notable example, have numerous holidays celebrating everything from independence



Day to the start of the World's series, between question, being a much younger industry, doesn't have a single one. To finish the crying need for some special legal holidays for aviation, we'll repeat that the first one set off for the official opening day of the pleasant shooting season. As the far corners of the flying fields are popular being operations for pleasure all pilots, mechanics, dispatchers and airport managers are out in full force at the crack of dawn. We've seen some fields where they were so thick they might have been mistaken for WPA workers except that they were moving around instead of leaning on shovels. We counted that that day might as well be declared an official aviation holiday as there's no work done anyway.



Boris Ananov

#### AFTER FAGG—WHAT?

It was with a real sense of regret that we heard of Fred Fagg's impending departure from the Bureau of Air Commerce. We have been following Bureau affairs rather closely over the past three or four years and have something of the closest between internal conditions today and what they were when Fagg took over from the former administrator. Not that he has been able to "wipe black and pass a miracle" to create the perfect government bureau in the space of a few short months. But out of the chaos of conflicting ambitions and petty jealousies that had been the rule rather than the exception when he selected the job he has succeeded in building up an *esprit de corps* that was entirely lacking before.

To be sure, the reorganization program announced last spring is not yet complete, but all those persons have shown a commendable willingness to work together as a team. It has been the Director's aim not only to coordinate the work of all his subordinates, but also to keep each department head fully informed of the problems and progress of other departments so that each feels that he is an integral part of the whole.

Naturally, all hasn't been "beer and skittles" for the Director. Undoubtedly Fagg has many times found himself blocked by political obstructions both from above and below that have tried him to the end. But we

doubt seriously that troubles of this sort were the primary reason for his decision to leave the Bureau. He is a man of too broad gauge to let such things get him down. After all, no one expected that he intended to make a career of the Directorate (even if such a thing were possible under our politically conscious regime), and, when a good opportunity came along, we should not be merely blind to his motives and ambitions, it was only natural that he should accept it. Fortunately, his new assignment will not take him away from the Bureau immediately, but will permit his remaining in Washington long enough to see through a number of jobs that he has already started. It is obviously his personal desire to hand over to his successor a smooth running and efficient organization.

Who that successor is to be, however, is still very much of a mystery. We feel that the administration will be making a very gross mistake if some positive action is not taken immediately. Certainly, by the first of the coming year a new Director should definitely be named, and he should function as right-hand man to Fugg during the months that he remains in office. If such action is not taken the administration will simply be duplicating the circumstances that surrounded the Vold appointment in 1932 and we will again be in for a period of political juggling that will seriously undo all the good work that Fugg has accomplished in the past year. Already a goodly lot of competitors in the field. Political questions are being strengthened, and nerves are being sharpened for as any of these things that may easily make the last Battle Royal look like a Methodist picnic. The Powers are present in it they will. Let them name a good man bothwith, so that the work of the Bureau may go on without any unnecessary interruption.

#### THE MARITIME COMMISSION REPORT

ON NOVEMBER 15 the long awaited report of the Maritime Commission was presented to the Congress. Of particular interest is the section on over-seas aviation proposed under the direction of Greener Looming. This where in this issue. On the aviation aspect will be found some of its details.

Three phases of the report seem to us to merit particular attention. In the first place, it is recommended that the way be left open for competition in the development of the highly important trans-Atlantic air routes. With all due respect to the great progress that has already been made in over-seas flying, the feeling is that the art is still so young that it would be detrimental to the public interest to restrict future development by the granting of monopolies of any sort at this time.

In the second place, we are glad to see that light-throat craft will probably not be excluded from consideration in the building up of our over-seas air transport. As we pointed out when the (Redundant) went down, we believe that it would be a definite mis-

take to overlook the possibilities of the airship in spite of the fact that most of the experience to date has been negative. At least these possibilities should be exhaustively explored before tossing them completely aside.

Finally, we can't overlook the fact that if the Maritime Commission's recommendation that it be granted supervisory over foreign air transport is acceptable to the Congress, we are faced with the introduction of another agency into an already over-complicated picture. To the evidence that has resulted from an uncoordinated effort of the Post Office, the Interstate Commerce Commission and the Department of Commerce to control American aviation must now be added another element, the Maritime Commission. That, certainly, won't tend to simplify matters any. But perhaps in the end it may prove to be a good thing. If too many fingers get into the pie it may occur to some one that the only sensible "out" after all, will be some sort of permanent federal commission to direct and to coordinate all our aviation effort.

#### A NEEDED REFORM

FOR SOME TIME past the National Aeronautic Association has been plugging for an idea that seems to us as extremely sensible that we wish to add our endorsement and to urge the industry to give the matter some serious thought and some active support. We refer to the proposal for the establishment of standing committees on civil aviation in both houses of Congress.

Too often in the past few years we have seen what happened to aviation efforts on the Hill. Matters that should be treated with respect by legislators with at least some rudimentary knowledge of what they are all about, get tossed around from one committee to another, usually ending up in the hands of someone who has discovered that there is no better way of making the front pages than by having his name associated with anything to do with aviation—no matter what.

The only way things of this sort can be avoided in the future will be by the appointment of permanent committees on aviation affairs made up of men who have the real interests of aviation development at heart and who have the time and the inclination to study the problem and to become reasonably expert themselves on matters pertaining to the air.

Practically all other phases of national affairs have their special committees in Congress and, certainly, aviation has become important enough in the national scheme to rate similar attention. Whether some sort of permanent interest will be just as futile to expect anything to come of the recommendations of the industry or of special committees of experts in the future as it has been in the past. The setting up of standing committees on civil aeronautics in both the Senate and the House of Representatives is a first and a highly important step toward the formulation of a sound national policy for aviation.



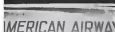
1 Donald Douglas does a little extra-curricular repair work to make the expression of the Stinson pilot—Wide World.

2 Fox-Bassett's John Cooper shows the French lines Made de Landings and Air France's Marine Division around the Ford Washington, L.L. International World.

3 Ray Brown this and a new (Baltimore) Over the Flying boat for in Baltimore—Wide World.

4 Right side of a C-47—Wash. radio turned on the first Boeing B-24 due out doors in December.

5 Joseph Douglas gets a new (Baltimore) Executive and the old (Baltimore) double.



## Camera's Eye on the News



# "Here Goes NOTHING!"

—a dangerous state of mind for the test pilot—or for anyone else who pushes a stick. Now flight testing has progressed from the dazed days into the realm of scientific research is told by

**Edmund T. Allen**

one of the country's foremost engineering pilots who is now in Seattle testing the Air Corps long awaited "Project A" (the Boeing XB-15). Recently he put the first of the Navy's Sikorsky "flying dreadnaughts" through its paces off Bridgeport.

EVERY FLIGHT WAS INTERESTING in the nature of a test flight. And it was a test typical of a period happily behind us, a period of knowing very little and guessing very much about how the airplane would behave once it got into the air under any given set of conditions. The test pilot put a new snarl of these days to rest when would happen. He told the engineer, "Well, here goes nothing," and was a bit surprised if his demands upon the craft were not safely. It was the era of the daredevil. The pilot was a kind of being different from all usual, thinking, planning creatures. He was not supposed to know what it was all about. Let the plotting vagabond figure their life coefficients, he was a man of the element "other," born to soar aloft and escape death by a hair, necessarily.

Many of the early accidents in testing can be traced directly to such lack of a careful attitude on the part of the test pilot, who usually relieved himself of responsibility by referring to himself as a "daredevil." In these early stages of the science of flying the pioneering spirit elevated the type of man who enjoyed himself in a field whose trust of the future were unknown. As the unknown gradually became progressively predictable, the increasing complexity forced a gradual upon such non-analytical pioneers. The attitude of being one who takes risks for pleasure was no longer possible because it obviously meant that one was not up to the job of coping unflinchingly with this complexity. Under the changing conditions of increasing knowledge, scientific risk-taking becomes a more responsible

Anyone embarked upon the business of designing and building experimental airplanes is usually an psychologically into a chance-taking world. It seems so much simpler and more

convenient of time to go ahead and try out the craft in the air to see what will happen rather than to reason out, test to design and experiment "on one's head." The development of the art of flight testing can be roughly divided into three periods or stages, which are psychologically if not always chronologically distinct. The first of these has been described as one where everything was unknown—there seemed to be no other way to get an airplane built and on the market than to try it out with as big a flourish as possible and on as many planes as the risks involved. There were no exceptions, as the careful Dredge and What Wright did a thousand times in much planning and experimenting as they did flying.

At the second stage of test flying development certain principles became

standardized in design, such as the angle in obtaining longitudinal stability. We no longer have to face a first flight with uncertainty as to whether the airplane will be stable or not. Stability of the design has been computed. A model of the airplane has been tested in a wind tunnel for stability and controllability. Finally a check has been made of the proportioning of the stabilizing elements in accordance with past test practice.

AT THE THIRD STAGE the properties of materials became more accurately known and methods of stress analysis in elaborated that exact yield points for wings could be predicted for each "specimen" as an airplane gets in peeling out quickly from high speed dives. The test program for flight tests

was a course of an elaborate determination of the stresses of control and stability quantities, performance, and cooling characteristics. The second period was characterized by an enormous enlargement in our store of data on the one hand, and an increasing tendency, on the other, to avoid further study by postulating by laboratory experiment, and by computation. The second period was even more a transition stage in the development of an attitude of test people than it is the product of chronological progress. And it brought a demand for the type of test pilot who was best on reducing his risks, and who was capable of a new interest in thought in planning for every emergency that might occur. Test pilots as well as designers were now expected to think about several nerves in a short game where each consider must shift the situation into a new pattern.

In the third stage there definitely emerged a new attitude toward flight testing, away from chance-taking of together and toward a complete analysis of each problem that can be test prior to flight. This stage is the second

perfect identity of the program in the second stage where a computer had to be chosen between the end of further correction and the end of insurance (also insurance), against the risk of "trying the thing out in the air." Those who are reaching for the attitude greatly decrease their risk by beginning their "testing" at the inception of the design rather than at the last minute when everyone's designer in production manager, it is to some that insurance is favored and mistakes difficult to avoid.

As was mentioned above, in the early stages the designer before all else tended to favor the attitude of daredevilry and to hold it in after the test pilot was already under way to become more scientifically analytical. Budget requirements frequently pushed the measurement of later days—as they measured it'll do today—back into the old pattern, to rely on a new attitude of testing and to get the aircraft into the air as soon as possible, leaving out its difficulties somewhere after the unbearable expense of finding out if it would fly

was over. The old Fokker organization was typical of this class of attitude, with Mr. Fokker laying out the design with drafts on the floor, flying the airplane the first test himself, then taking and flying and he was attacked with the "test" of the course. Some of the early airplane designs were laid out solely in this manner, Harry Houser taking the risks that were cheaper than a mathematical department. Occasionally they would lose an airplane and the test with a new design which turned out "in the air" to be so bad-better and to have such inadequate development, that even with the double check nothing could be done to keep the airplane from falling at once. Houser, doing all his testing without a parachute at that time, attempted half a dozen times to land the airplane, got the wheels on the ground at high speed or with the engine fully on, but could not slow up or cut the throttle without having the airplane "come." Finally he went off a tree so as to let it with one wing and out the outside just as he hit. Fortunately he survived. The relation between cost of gravity location and stability was known at the time, and the accident, like so many others, was not owing to lack of knowledge but to lack of care.

It is, however, precisely the mark of scientific progress when these afterthoughts become an subsequent truth post-thought. It is usually easy to see after any accident how it could have been avoided, how it should have been foreseen. There are very few accidents any longer in aircraft trials.

STRESSING is a typical problem dealt with characteristically in each of the three stages. The typical method of these still belonging to the naive stage was simply to put the airplane into a long climb, possibly on the first flight, and when the stress arrived for descent recovery, say after ten miles, to light each way out as best one could, if one could. The second stage type of spin testing began with a determination of the factors entering into spin recovery, such as unbalanced vertical tail surface, effective control surfaces, distribution of weight along the axis of the airplane, and other less apparent factors. The provision of an adequate, carefully designed spin-check was an essential at this stage. Any test pilot who attempts spin recovery on a machine without a spin-check will be

(Continued on Page 12)



The spin-check is often a life saver for both plane and pilot. The above illustration is a type used by the R.C.A.F. flight test section in its exhaustive study of spinning.

# MILAN

## Shots



Conspicuously absent from last year's aircraft show in Paris were Italy and Germany. This year at the second Milan Aero Exhibition Hitler's Swastika and Mussolini's Fences practically stole the show. Scattered among familiar types, AVIATION'S correspondent at Milan spotted a number of new and interesting designs.

### ITALY

1. In the medium bombing class, the Breda 40 with a total of 2,000 hp. from two Fiat engines. Span is 40 m. max. h. is 10,000 m. max. speed 350 km. per hour.

2. The Fiat 5-000 by Montecatini, a development from Fiat's single-engine biplane. It is built largely of wood. Fuselage is of monocoque type; wings have three ribs across the middle span of 30 m. max. h. is 10,000 m. max. speed 350 km. per hour.

3. The Fiat 5-000 by Montecatini is built entirely of wood. Span is 30 m. max. h. is 10,000 m. max. speed 350 km. per hour.

4. The Standard 1-8 resembles the American Stearman-Barnhart. Span is 21 m. max. h. is 10,000 m. max. speed 350 km. per hour.

5. Heinkel exhibited the H.S. 1 "Hindler" or General type, all wood, with a 100 hp. E.H.B. engine in the nose. It is built in Germany. Span is 18 m. max. h. is 10,000 m. max. speed 350 km. per hour.



### GERMANY

6. The Junkers Ju 50 is powered with two BMW engines of 100 hp. each and carries three machine guns forward. Span is 18 m. max. h. is 10,000 m. max. speed 350 km. per hour.

7. A prototype of the latest design for the Ju 50 is shown.

8. The Heinkel He 111 single-engine fighter carries two machine guns in the fuselage and two cannons in the wings. Power plant is a 100 hp. supercharged engine.

9. Detail of the Heinkel He 111 fuselage shows the engine, fuselage, and tail section.

10. Designed for landing in the German desert, the Heinkel He 111 is built in Germany. Span is 21 m. max. h. is 10,000 m. max. speed 350 km. per hour.



Photographs and specifications by Fritz Winkler, Berlin.





Fig. 1.  
Earth and celestial spheres show  
key altitude, latitude and the  
celestial triangle.

By  
**Lt. Comdr.  
P. V. H. Weems**  
U. S. N. Retired

sidered, but the latest practice is to consider only the *indicated* triangle on the chart, thereby simplifying the solution.

The reader will note at once that the equator, poles, meridian, etc., on the earth have their counterparts in the celestial sphere. Note carefully that the position of the sun or other celestial body may be projected to the surface of the earth and when so projected, the corresponding position on the earth is called the *Geographical Position (G.P.)* of the body.

Another important definition is that 1° of arc on the Earth is a geographical mile (6080.27 ft.), and in navigation, to find a position by celestial bodies, it is necessary to know the G.P. of these bodies at any instant of time. These data are found in the *Nautical Almanac* (or *New Nautical Almanac*). The *latitude* (called *Declination*) and *longitude* (called *Greenwich hour angle*) of the G.P. of all bodies used are tabulated in the *Almanac* for suitable intervals of time, and the exact values for any desired instant found by interpolation. We are now ready to take up—

#### Celestial Navigation Theory

An observer at any point on the earth's surface, except the poles, sees the altitude of any heavenly body change with time. The altitude can be calculated for any instant of time by methods to be described later.

The altitude of a heavenly body also changes with a change in the position of the observer. The amount of the change depends on the direction in which the observer moves relative to the geographical position (G.P.) of the observed body. If he moves *due* mile directly toward the observed body, the altitude is *increased* by 1° if he moves *away* by 1°; *while* if he moves in any other direction the altitude is *changed* by a predictable amount.



The observer is also free, usually, that, if the altitude of heavenly body is known, the distance from the geographical position of the body to the center of observation is known. Celestial navigation is based on the application of this principle. Therefore, we may say the work of the navigator in celestial navigation consists in finding his geographical coordinates of latitude and longitude by bearing *back* with reference to the geographical position of one or more celestial bodies.

The exact position of a body, in terms of declination and Greenwich hour angle, may be taken from the *Almanac* for any instant of time, and this position may be plotted on a globe or chart as latitude and longitude. A circle with this position as center, and with a radius equal to the observer's distance from the position as determined by the sextant (i.e. 90°-altitude), would be the observer's circle of position; the observer's position must be somewhere on this circle. If two bodies were observed, the intersections of the circles of position would determine the position (or two possible positions) of the observer.

Since the circle of position as determined above would be very large, a small arc of the circle may be considered as a straight line, at some point on which is the observer's position. If it were possible to lay off on a chart with sufficient accuracy the distance of the observer from the geographical position of the body

practically but the desired effect is secured by the means about to be described.

The navigation triangle in Fig. 2, which is a representation of part of the earth's surface, P is the desired pole, Z the position of the observer and S the geographical position of a star. These three points form the vertices of a spherical triangle ZPS which is known in the astronomical or navigational triangle. The side PS is the polar distance (90°-declination) of the star, ZS is the earth distance (90°-altitude) of the star, and PS is the coaltitude (90°-latitude) of the observer. The angle ZPS at the pole is the local hour angle, and the angle PZS at the zenith is the zenith or direction of the star. The angle PSZ at S, which is seldom used, is called the *parallelic angle*.

*Solving the astronomical triangle*—The question from which the astronomical triangle is usually solved are the sides PS and ZS, and the indicated angle at P. The assumed latitude subtracted from 90° gives the side PS. The declination subtracted from 90° gives the side ZS. The angle at P is the difference between the right ascension of the body (geographical position) and the assumed longitude of the observer.

By specially arranged tables the latitude distance and local hour angle are used directly in solving the triangle for the computed altitude, which is 90° minus the side ZS, and (Two in page 74)

## AIR NAVIGATION Finding Your Way in the Air

### IV. CELESTIAL NAVIGATION

**C**ELESTIAL NAVIGATION is the method of determining position by means of celestial bodies. While this method has been in use for centuries, its application to air navigation is of recent origin, dating back only about twenty years. In this short time, however, and due largely to the development of Air Navigation, both the interest and the methods have undergone radical changes. While space does not permit a detailed exposition of the method, it is the intent of this and the succeeding installments to explain the principles on which the method is based, and to indicate in a general way how it may be accomplished. The reader

may then judge for himself whether or not to take up the practical application of the method.

In this discussion, certain terms new to the beginner, but no more mysterious than such terms as "latitude" and "longitude", will have to be introduced. In defining these terms, principal use will be made of illustrations and a dependence on the common sense of the reader.

Let us now introduce the *speed terms* used by a celestial body of Fig. 1 which shows the earth and the celestial sphere with the navigation triangle outlined and with the essential definitions indicated. Particularly, the celestial triangle was con-

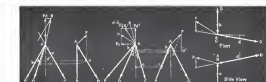


Fig. 2. Navigation triangle and circles of equal altitude.

# Solving SPACE

# STRUCTURES Graphically

## Part I By Elias Moness



**T**HE APPLICATION of graphical methods to the solution of space structures, i.e., of structures all members of which do not lie in the same plane, is difficult and often impossible. Yet, the graphical method is so much quicker and easier than the analytical method that it is very desirable to extend its use to space structures as well. It is the object of this article to demonstrate a method for solving any statically determinate space structure graphically.

Let us analyze the simplest of all space structures—a rigid body as is used for finding centers of parallel forces.

### Case 1

Fig. 1 shows such a trapezoid with the load "P" applied at the apex. Note that two views are required to show the structure and the applied load. Resolve load "P" into two components,  $P_v$  in the plane of members AB and AC, and  $P_h$  normal to this plane. Since AB and AC cannot take any load which is perpendicular to their plane, it is obvious that  $P_h$  is taken entirely by AD, since  $P_h$  and AD are in the same plane.  $P_v$  is the horizontal component of the stress in AD, hence, we project  $P_v$  on AD and find AD to be the stress in AD; this can be solved using the same scale as the one to which "P" was drawn. Next, we mark the nature of the stress in AD by means of arrows (compression).

The members AB and AC must take one of the load  $P_v$  and also of the vertical component of  $P_h$ , hence we lay off a vector equal to  $P_v$  (vertical component of stress AD) as shown and find the stresses in AB and AC in the usual manner. Finally, we mark the nature of the stresses (tension) and scale off their magnitudes.

An examination of Fig. 1 will show that instead of resolving "P" into components, we could have solved the side view as if it were a plane

The graphical method outlined in this series of articles saves three-quarters of the time required for computation and the results are accurate to within one percent of computed values. It has been used extensively by the Douglas Engineering staff and can be applied generally to any space framework



structure consisting of two members only. (See Fig. 1a.) AD is found by drawing PD parallel to ABC, and has, of course, the same value, as before; the phantom member ABC has a tension equal to PD; hence, we apply a tension load PD in the structure in the other view and solve; again we mark the nature of the stress and scale.

Note that the original applied load in the front view is ignored when we apply the phantom load PD. Care must be taken to apply a tension load, if the phantom member ABC is in tension.

The case just analyzed is a special one:

1. The load "P" is in the plane of the paper in side view.

2. The structure is symmetrical at the front view.

3. AB and AC are in the plane of the paper (side view is vertical).

4. AD is also in the plane of the paper.

5. AB and AC are superimposed on each other in side view.

### Case 2

In Fig. 2 condition 1 is dissatisfied by using a space load, i.e., by giving the load "P" a side component. To solve this case we proceed as before and apply PD in the front view. At this point we must take into consideration the side component  $P_h$ . This we do by resolving PD with  $P_h$  in the front view. This gives the result-

ant R. We then proceed as usual.

In Fig. 3 condition 2 is dissatisfied by making the structure unsymmetrical in front view. The method of solution is the same as above. Note that only the side component of P in the front view is combined with the phantom load and not the whole load P. This is so because the vertical component of P has already been taken care of in the side view.

### Case 3

In Fig. 3 condition 3 is dissatisfied by resolving the plane of AB and AC, so that they are no longer in the plane of the paper in any view (AD is still in the plane of the paper in side view). We obtain AD by drawing through P a line parallel to ABC. But, since the front view shows only the vertical projection of AB and AC, we must apply not the entire phantom load PD, but only its vertical projection PD'. Solving then the front view in the usual manner, we obtain not the true length of the stress vectors, but only their vertical projection. These vectors cannot be scaled directly. We must resolve them into the plane of the paper and after thus finding their true length, we can scale them.

The complication introduced by the fact that the plane of AB and AC is not vertical, in that we must deal with projections, and finally find true lengths by resolution of the vectors.

Fig. 3a shows the same case for a space load and an unsymmetrical structure. Note that both stress vectors had to be resolved since they are of unequal length.

### Case 4

In Fig. 4 condition 4 is dissatisfied by drawing member AD away from the vertical plane. Now even if the members of the trapezoid lie in one plane (See page 20).



# Producing

# the Ryan S-C

By T. Claude Ryan  
President Ryan Aeronautical Company

TOP PHOTO BY  
C. F. McFARLAND  
AVIATION



How manufacturing methods have been developed in conjunction with design to simplify metal plane production problems through utilization of the drop-hammer forming process.

quantities that have been customary in the past.

Although designers have for years argued with the idea of aluminum stamped all at once in one large piece, or poured out of some synthetic material in a single molding operation, or somehow interested in us to avoid the tedious hand fitting of countless numbers of small parts. We believe that in the development of the new S-C we have approached this steel of the one-piece airplane.

It all started with the development of the S-T metal-banded sport trainer which went into production in 1934. Four years of manufacturing experience with metal gliders, plus the steadily increasing demand for the same period of a wide variety of metal parts on contract for other airplane manufacturers, have given us the opportunity of fully developing the application of metal construction to the small airplane.

Our basic effort was to develop a design in which the number of individual parts was held to the lowest possible, the elimination of hand fitting, and the simplifying of cutting and costly assembly operations. It is a much more difficult assignment to simplify a structure than to "complexify" it. But by extensive study, experiment, and aid of ingenuity we have gradually

evaluated the design of the S-C to the point where, as it is now in production, we feel it is the most simple manufactured metal airplane, and the most adaptable of any yet developed to an efficient production system in fairly large quantities.

Such simplification and elimination of hand fitting requires precise manufacturing. Precision manufacturing of parts made of aluminum alloy sheet could be economically achieved in only one way—by stamping them all out with the same die, and that is essentially what we have done.

The Ryan S-C cabin plate is almost entirely of 24ST Alclad aluminum alloy sheet construction. And every part of it, with few exceptions, is either a flat sheet, or a drop hammer stamped. Flat sheets are all stacked and drilled from master steel templates, and stamped sheets are all formed from dies which accurately reproduce a given part exactly over and over again.

In the S-C design we have reduced the total number of parts in the plane, reduced the amount of fitting required to assemble it, eliminated completely most of the costly parts originally required, and made layout and alignment free in assembly operation unnecessary.

The S-C fuselage closely follows S-T practice, the forward portion being assembled on a provision jig, while formed alloy sheets are riveted to bulkhead bulkheads which to the

landing structure in the main wing spar and engine mount. As on the S-T, the alloy portion of the fuselage is a simple monocoque structure of one piece, shaped from flat alloy sheets without compound curves. These sheets are easily formed to the shape of the fuselage after layout of the sheets from master templates. The universal floppy sheet metal cone is then developed into a sturdy fuselage structure by riveting the bulkheads in place. This is all done without recourse to any jig, as the layout from master steel templates assures perfect alignment, although all rivet holes are drilled before final setting of bulkheads in the sheet.

From the cabin rearward, the fuselage is a true monocoque structure, and is assembled to the cabin section in a master jig, all other operations on the fuselage structure assembly being performed with assembly jigging dimensions.

The wing is similarly simplified to obtain the maximum advantage from the precision drop-hammer die-stamping process. With a monomer type wing structure, each panel is attached at the fuselage by means of only three lower bolts. The word "monocoque" now is a development of one (This is page 82)



Showing part of the history of four drop hammers on which most of the Ryan S-C parts are produced.

Master planer model of Ryan S-C cabin plate being used for development of form by dies to produce bulkheads and fuselage structure after sheets to be formed on the drop hammer.



Master fuselage in its cabin portion in development. Same portion before attached to second section from the die. Completed fuselage structure in background.



Automatically moved load in Ryan S-C cabin portion.





## More about that **WACO 3-Wheeler**

High Perry discusses some of the advantages of three-wheeling and reveals a few details of construction and performance.

ONE OF THE obvious advantages of any three wheel gear is, of course, the excellent standing on the ground at all times. At no time is take-off or landing a valuable sacrifice for so instant. Next, of course, is the maneuverability of the airplane of take-off and landing for a seaway. While taking off the pilot remains in a level with the ground position (as he would if taking down the runway in a motor car), until the airplane is ready to leave the ground. In landing, large flaps (such as those fitted on the Waco N) permit him to approach the very edge of the flying field and, just as the airplane descends toward the spot he wishes it is practically impossible even for the novice, to build up dangerously excessive speed in this manner. As he approaches the ground it is natural to decrease his angle of approach and level off, but he does not have to put the airplane in an attitude beyond level flight with the nose pointing upward. On the contrary, he maintains the natural level with the ground attitude at which he took off.

In leveling off above the ground the weight acts as approximately zero pressure and there is no tendency for the ship to bounce back in the air and the handiness of the ship is completely absorbed by long slow wheel travel of the three wheels. If the front wheel hits first while there is a slight bounce, again there is no positive resistance and the airplane bounces only to level position and settle on all three wheels. Lastly, if (usually by an experienced pilot) the rear wheels are put

down first and an old style conventional three point landing is made, the ship rocks forward at once without major shock onto the third from wheel.

Another obvious advantage of this gear is, of course, the stability of the ship in level or turbulent air. It is doubtful that the directional stability of any three-wheeler on the ground is good at all times. This is achieved with a gear with the front wheel fixed so roller and one steerable. Above speeds of fifteen to twenty miles an hour the airplane cannot be turned sharply, so much of a positive effort is made. Ground looping is, therefore, completely eliminated and winged loadings and take-offs present no problems. Below these speeds steering may be accomplished with ease.



Continued



The Model N-8 Waco follows quite closely the conventional Waco design of the late 1920s series. The fuselage is of welded aircraft tubing with a conventional motor mount, the structure is wood-banded for shape and fiber covered.

The wings are of wood construction with spars of aircraft spruce and built-up ribs of spruce with plywood plywood panels, the ribs are separated with duralumin compression tubes and double braced with three internal tie rods, wing tips are finished to a nice bevel edge. Ailerons are of the same design and are provided an upper wing only. Flaps of duralumin are provided and are vacuum operated from a single tank located in the lower wing covering transmission action on all surfaces. The entire wing structure except flaps is fabric covered.

Horizontal and vertical stabilizer surfaces are likewise of wood structure, the horizontal stabilizer being first covered with plywood and both ends then fabric covered. Elevators and rudder are of welded aircraft tubing fabric covered.

The cabin interior is finished with leather headliner. The rear seat is solid with individual belts for two. The front seats are individual and are adjustable. Coast situation has been given the design of all seats to ensure comfort on long trips.

Strike pods operating overseas by (Turn to page 37)



## THE World's Greatest Ocean Transport

ANSWERS THE CHALLENGE OF THE ATLANTIC

But ten short years ago all the world drifted to the amazing news of Charles A. Lindbergh's non-stop flight, New York to Paris. Why has the world waited now to see an oceanic transport capable of carrying commercially profitable pay loads of passengers and cargo non-stop over the trans-Atlantic route?

The Lindbergh flight was a magnificent accomplishment. It was a triumph of audacious experimental flight engineering with a great ship, a great engine and a great pilot—all wisely balanced to make a reality a long-sought dream. That flight pioneered. It carried no pay load, no cargo but a single man of less than 200 lbs. In 1927 an amazing achievement!

Before this historic flight could be made the basis of a commercially practical reality, there were great areas of engineering to be developed, great fields of experimental discovery to be explored. Spanning the Atlantic non-stop with suitable pay loads of passengers and cargo—here was a challenge to the aviation engineers of all the world. It meant the development of new designs, new advanced structures, new over-all efficiency, to provide for pay loads large enough to make non-stop Atlantic service commercially as well as aeronautically successful.

With the proven design of the remarkable Martin-built Chaco Clippers at the first step in ocean transport pioneering, a new great ship has been conceived—a ship that can fly the Atlantic non-stop—quickly, easily, safely and economically. In this great new Martin over-ocean transport, one long-range efficiency has been forged into reality—a ship capable of non-stop trans-Atlantic service (3,500 more miles—5,000 miles in still air) with large and practical pay loads of passengers, mail and cargo. Here is another significant milestone in the boldest history of aircraft development.

It would be easily possible to build a seemingly comparable ship with greater cruising range, or greater total pay load capacity, or greater speed, or greater hull dimensions. But nowhere in aviation, here or elsewhere in the world, has such an amazing profit been made, measured to that fine balance of all the factors necessary for non-stop air service across the Atlantic on a commercially practical basis.

This great ship will be delivered to its foreign purchaser shortly. To those American operators who are interested in establishing non-stop trans-Atlantic service, we offer this design, ready for immediate construction.

THE GLENN L. MARTIN COMPANY

BALTIMORE, MARYLAND, U.S.A.

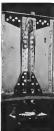
Builders of Dependable



Aircraft Since 1909

# Martin Ocean

Wing structure  
before covering



- 1 Wing beam spine and stiffening ribs at center of ship
- 2 Center rib rib
- 3 Alternate structure shows ribs ribs. Topper between beams to actively serve by the leading edge box structure
- 4 Draft outline of the low aspect, the approach to the section line
- 5 Fly control structure on rear beam



The entire structure is paid-off in stainless steel money metal, that is beginning to be in track at all for the follow which makes five or six hundred mile range between stops. But the apparatus which has to fly across an ocean under the full largest range in gradient and the number of miles per passenger quickly reaches appalling proportions. After a while it begins to get on his nerves. The ship he had to dream about gasoline and the frequent refueling, or to dream about the cost with sustained specific fuel consumption figures.

To this problem the Glenn H. Martin Company has devoted much time and research. The goal is commercially profitable trans-oceanic transportation by air. To approach that goal Martin has considered all of the practical problems in the light of their past experience. They have developed and built an aircraft that will support itself as well on the amount fuel as it does in the air.

The first job was to fly long distances without stop over water with regularity and reliability. This has long since been accomplished and is now being done with Martin equipment. The next job was to fly over the great long distance with a sufficient load of passengers or freight to pay some profit to the stockholders. This too has been done with the most famous Martin products—the China Clippers. But the final job is to increase both the range and payload. That has been done in the new Martin ocean transport, Model 116.

Although the new Glenn Transport represents an important forward step in large flying boat development, it is most conservatively described in terms of its characteristics, the China Clippers. All of the features that have withstood the tests of time and service have been retained and many notable improvements have

# Transport

16,000 lbs. of payload in a long range flying boat

have been added. The foremost of design contributing to operating economy and load carrying capacity have been substantially reduced.

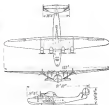
The most striking improvement of relative efficiency lies in the respective payload of the two Martin models over the same route. From Sea Princess to Honolulu, the China Clipper carries a payload capacity of about 2,500 lb. The new Martin Ocean Transport on that route can carry in excess of 16,000 lb. payload—and at high speed. Thus, the next step in our ocean transportation boat takes.

Generally the new ship resembles its predecessor in that it is a balanced monoplane flying boat with inherent stability when alone provided by Martin wingings. The most striking difference in appearance is furnished by the tail section. The departure from previous practice improves control in flight, particularly with sea engine stopped.

Modifications in the hull are few but are of great importance. No changes have been made in the general construction and the passenger accommodations are the same

excepting an increase in the number of sleeping berths from 18 to 26 and the addition of an additional toilet compartment. The hull is no longer covered in integral sections between the cabin floor and hull bottom. It has been moved outside to the wingings, leaving passengers free from the possibility of future penetrating into the cabin space.

Hull lines in general are similar to that of the earlier ship, but up to the bow the shape flare has been relieved with no dimensional effect indicated in water tests and with definitely reduced drag. A weight saving has been effected by the substitution of the Martin type of anchors. Another important modification in hull form is found in the tail. The disk sides of the Clipper have been replaced by curved surfaces resulting in an efficient section increasing the space for storage of light materials and providing two crew berths instead of one. One of the most valuable features of the new construction was to the operator is that it makes it possible for a man to go back to the tail tip where he can reach the steering helix



located there or inspect the structure and controls at the other perpendicular.

Wingwing of the cabin structure forward has greatly improved the accommodations for the flight engineer. The position of the two pilot seats has not been changed but the cabin space has been moved down into the lower level again just forward of the ladder leading up to the pilot's compartment. This leaves space for a navigator's post complete with chart board, compass, and observation window in the forward portion of the cabin. Further back is the office of the flight engineer with its private engine hatch on the port side and a compact but complete instrument panel to the starboard. The flight engineer's seat is mounted on a track and can be moved back and forth in the extreme rear of the cabin. It is added to his other duties that office is in a strategic spot to inspect the control cables which pass out of the release line on external drag at a point just ahead of his seat. Thus the control cables bypass the passenger cabin and their examination is possible without disturbing the functioning of the interior. They return to the interior of the hull at the rear compartment.

Changes in the wing are more aerodynamic than structural and it is here that the remarkable improvement in efficiency are obtained. The increase in aspect ratio from 7.85 to 10.75, the shore approach to elliptical plan form, a change in swept section from a modified Göttingen to the NACA 23000 series, and the addition of flaps are the essential differences between the old and new models. Because of the larger span



Structure of hull showing the new elliptical section

(Part in page 42)

# DELTA AIR LINES

# *"Tops in Travel"*



DELTA, the True-Southern Route through Dixie, is known as the short cut between Florida and California. Delta serves southern hospitality at its height. Truly, "Tops in Travel."

## *across*

# DIXIE!

**A**LWAYS maintaining the highest standards of luxurious air-travel, DELTA AIR LINES now add still another safeguard... their Lockheed Electras are lubricated with New Texaco Airplane Oil.

This remarkable oil has surrounded the entire aviation industry. Test engines lubricated with it have come through the most severe tests entirely unharmed. Through the use of this oil, wear has at last been reduced to a new low.

Trained aviation engineers are available for consultation on the selection and application of Texaco Aviation Products. Prompt deliveries assured through 2670 warehouse plants throughout the U. S.

The Texas Company, Aviation Division, 135 East 42nd Street, New York City.

TEXACO AVIATION GRADE • NEW TEXACO AIRPLANE OIL FOR ENGINE CYLINDERS AND BEARING DISCHARGES BEARING AND DRIVE SEALS • TEXACO CAPILLA OIL FOR MAGNETO AND PLASTER BEARING • TEXACO GRADE No. 2 FOR ROCKS KING AND WHEEL BEARING



# TEXACO

AVIATION  
December, 1937

31

# *Aviation* PRODUCTS

AVIATION  
December, 1937

32



Bolt bolted in place of wing attachment

and the elevator from there is less damping in yaw and better control is obtained. Ailerons are much smaller than those of the Chin Clipper. Tabs and servo controls are provided but may not be necessary because of the generally improved control.

An important feature of the specially designed flap is the "Auto-dog" shutter. The opened shutter shows the flap because the lower surface of wing end flap when the latter is in normal position. In the first degree of downward flap travel the shutter moves upward, completely opening the gap.

A number of refinements have been made in the wing structure but it is of the same general type as that of the earlier model. Detailed changes in the rib construction are shown in the accompanying illustration. A new feature is the variable camberments in each wing tip useful in the remote event of damage to the wingtips. Covering is integrated on top and on the bottom with deck riveting used only at the reinforced leading edge areas.

The coverings have been cleaned up considerably. Their bottom surfaces are the same as in the earlier model and are varnished. The top surface, however, has been treated to approximate the 25,000 series of alodine. Basic research directed the changes that have been made in vertical position, dihedral, incidence and span. Basic research has also produced turbo, out of 175 gal/hr capacity, the other of 1,200 gal

From there the fuel is pumped into two 300 gallon service tanks in the wings. Accessibility of the fuel and waste gauges has been improved by moving them up out of the belly into a forward compartment to the right of the pylon.

Wing tunnel tests on a scaled-up version of the Chin Clipper tail showed the desirability of improving the twisted characteristics to the Martin engineering staff and about to design as fast from tail and upper-



See wing structure providing later G-WC tanks of 1150 gal fuel capacity



Detailed view of wing rib construction showing the web box girders that serve as baffles for the integral fuel tanks

only it has turned out very satisfactory.

Learned under Red patents, the new tail is of particularly clean design. Its normally high aspect ratio is increased as effectiveness by the end plate rollers. The vertical area has been increased without any movement in horizontal area but the efficiency of the horizontal tail is much improved. Invaluable for maintenance and operation are the removable steps in the fin and landing gear door leading edge of the horizontal surface.

A good many engineering hours could be spent in designing for the installation of two G-2 Cyclones in a flying boat of this type and the complexities that might result were foreseen by Martin engineers far in advance of the initiation of construction. Accordingly, a test model was built so that the details could be worked out between the engineering department and the factory. The result of a good many hours of testing in the test model has been one of the cleanest power plant installations we have seen not only from the standpoint of original installation but of operating maintenance. Fit from the customary clutter of the small space behind the engine, the installation gives one the immediate impression that it would be very much easier to work on than the average automobile.

To improve the general accessibility of the engine, extensions have been added to the right folding service ducts in the leading edges of the wings. These extensions are completely detachable and are used only when it is necessary to get at the front end of the engine or the propellers. However, it made for the attachment of long leads to the wing space for removal of propellers or heavy parts of the power plant.

The propeller plane has been moved forward to reduce interference of the wings and 12 foot, double-bladed Hamilton Standard constant speed type with quick bushes and shaper rings are used. Shaper rings are a modification of the bomber type with NACA cooling and Vought type cord flaps. A flap angle of 45 deg forward was found better than one of 20 deg for the cord. Soler exhaust manifolds and Lord mountings have been used in the engine installation. An entirely new application for the Com of filter has been found in the hydraulic system.

Because of its properties the

(Continued on page 82)

# Experience

## THE PRIME REQUIREMENT FOR DEPENDABILITY



THERE is no substitute for experience in flying or in making wire products. As experience develops the expert, trustworthy pilot, our inspection of more than one hundred years of wire-making has taught us how to make wire that is best suited for its job—dependable in performance. We know the strength and flexibility requirements of airplane strand and cord, and know how to make wire products that meet these needs.

Our 7 x 19 Wire Construction, a semi-rigid galvanized cable of high tensile strength, is designed to withstand sudden stresses and excessive vibration.

Strength and flexibility are com-

bined with the advantage of a small diameter in the 7 x 19 airplane cord. In this construction, six strands of seven wires each are wound around a core of one strand of smaller construction.

Our 7 x 19 cord is extra flexible. A special grade of steel is used in this cord to give it high tensile strength. Its construction consists of six strands of stainless steel each around a center strand of stainless steel.

These products have proven their worthiness in actual service. They meet all of the requirements of the latest U. S. Army and Navy Aeronautics specifications. They are furnished in either galvanized, or USS 18-8 Stainless Steel.

## U-S-S AIRPLANE STRAND AND CORDS

AMERICAN STEEL & WIRE COMPANY

Cleveland, Chicago and New York



Cable Steel Company, San Francisco, Pacific Coast Branches—United States Steel Products Company, New York, Export Agencies

# UNITED STATES STEEL

## NEW BAROMETRIC SCALE

for the  
KOLLSMAN  
SENSITIVE  
ALTIMETER



WINDOW PLACED  
WHERE PILOT'S  
HAND DOES NOT  
HIDE THE FIGURES

Kollsman—the world's first and finest Sensitive Altimeter—now has a "high visibility" barometric scale. Located in a window at the right of the dial, the large figures are well above the pilot's hand on the setting knob, and so may be read easily and not precisely. Write for detailed specifications and full information on the Sensitive Altimeter and other Kollsman Precision Instruments.

# KOLLSMAN

PRECISION AIRCRAFT INSTRUMENTS

KOLLSMAN INSTRUMENT CO., INC., 1 JUNIUS STREET, BROOKLYN, N. Y.  
WESTERN BRANCH: GRAND CENTRAL AIR TERMINAL, CHICAGO, ILL.



## Two New Cubs

Sport and Trainer Models embody  
Many Improvements



### Specifications

Length overall	37' 1"
Height overall	6' 8"
Wing span	37' 1"
Total area	175 sq. ft.
Chord	5' 2"
Weight empty	114 lb.
Useful load	450 lb.
Maximum allowance	20 lb.
Cruising speed	55 mph
Top speed	60 mph

**M**ANUFACTURERS OF THE FINEST flying equipment in the world, Kollsman Inc., is proud to announce the introduction of two new models to be known as the Cub Sport and Cub Trainer. Although basically the same as the J-2, it is outwardly appearance the new Cub Sport reveals more than fifty major and minor improvements, within and without. The exterior has been greatly changed by the addition of a new radiator and fin which add greatly to the directional stability of the ship. The fin has been more than doubled in size and turned into the fuselage. Construction of the "Sport" is the same, three tube fuselage with fabric covering and spruce spars with metal ribs for the wing are the same as before. Whereas before the Cub was made a cabin flap by adding engineers, the other of the Sport is an integral part of the fuselage, a living complete window proving of windshield and window. The cabin itself is completely upholstered on sides top and rear and is backed by doped fabric to prevent stretching of the ship's end cloth. Except for the radiator cables, all other control wires are now beneath the floor.

The length of the cabin has been increased seven inches allowing substantially more leg room for occupants of both seats. The rear seat, which is of hammock type, extends the full 36 in. width of the cabin. The front seat made up of upholstered fabric is wider and deeper. Both seats are deeply upholstered according to latest automobile specifications.

The stick/rudder control mechanism is now actuated by a cable on the left side of the stick. The wire cable

which connects with the stick/rudder may be adjusted for tension by a push up puller accessible through an inspection window. The instrument panel has a control switch which allows the selection of extra instruments without cutting into the panel itself. When stopped, compass or both are desired, as insert with one or two holes supplied replaces the control part of the panel. Gasoline shut off and ignition switch are located in the upholstery on the left side. The left hand window slides up and down to reveal the upholstery.

Engine allowance on the new ship which has been issued a separate ATC is rated to 20 hp with an overall gross load of 1,000 lb. Wires are used for navigation lights, and other bearing wheels and load into which are standard. Different bearing of the top deck, and addition of struts on the side of the fuselage mutually increase the speed of the Sport which is available with either the dual or single ignition Continental A-40.

The Cub Trainer is basically the same as the 1917 J-2. Standard equipment now includes radiators, oiler bearing wheels and over-sized tires.



Improved interior of the new Cub Sport

# Aircraft Radio

More Equipment for Communication and Navigation by Resney and Fink

## Bendix Announces Two New Receivers

for Communication and Beacon Use

Wata covers: Iron, W. P. Ellstrand, Bendix was provided in change of the planning of Chicago, of two new aircraft receivers, the RA-2 (intended for communication use, and the RA-4, for beacon signals, also a direct-coupling autotune loop.

The RA-4 Beacon Receiver is an 8-tube unit, using one r-f stage and two i-f (150 kc.) stages, covering the range from 190 to 415 kc. continuously. A relay-operated tuning switch is built into the receiver for the purpose of automatically tuning the set to the desired frequency at 250 kc., by simply throwing the control switch. Manual sensitivity control and automatic volume control are provided. The receiver has two entirely separate audio channels, so that either at two pilots may use the receiver independently. Two-card indicator coils are used in both r-f and i-f circuits.

The beacon receiver weighs only 14½ pounds with tuning dials and knobs, a low weight made possible in part by the use of TST dual chassis, panel and cabinet. All the electrical connections, including antenna and power supply, are made through a

single plug on the front of the receiver. The power supply required is 1.2 amps at 12 volts and 45 milliamperes at 250 volts. A feature of the receiver is the flexible shaft tuning device, which operates through a 364-inch reduction gear. This gear contains a rotatable drive-head at the panel which can be rotated in steps of 30 degrees to accommodate the angle at which the flexible shaft approaches the set. This provides better and simpler installation. Dual base tubes are used (type 6K7G r-f, 6X25G oscillator, 6L7G first detector, 2-6K7G i-f, 6K7G second detector, 6X4 and 6X5, and 2-6X6G double audio output). The receiver will deliver 500 milliwatts from a signal of 1 to 2 microvolts, and provides a selectivity 50 db down at a bandwidth of 9.5 kc. at 200 kc., 30 db down at a bandwidth of 11.2 kc. at 400 kc.

The communication receiver, type RA-2, is designed for voice communications in the band from 300 to 7000 kc., in which any 4-presentation frequency may be chosen. The set is a 18-tube superheterodyne, the last-up being the same as that of the RA-4.

except the 6K7G tube is replaced by a 6X6G and a 6C7G, and that a 6K7G is used as an audio amplifier.

The tuning circuits employ air-dielectric trimmer capacitors, which are actuated by the same shaft as the coils. Each antenna coil is provided with five taps which allows accurate matching to the antenna at each of the eight pre-selected frequencies. The r-f stage is double-tuned, while the audio section is designed to operate crystal-controlled, using low-temperature coefficient units. An untuned coil, used in conjunction with each crystal is placed in the cathode circuit of the oscillator tube. Low-carbon substance coils are used throughout the r-f and i-f systems.

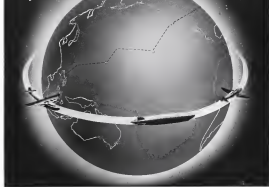
The a-c system, of the delayed amplified type, will hold the output current within 5 db, while the input varies from 1 to 100,000 microvolts. The manual sensitivity control is applied to all but one of the a-c controlled tubes. This latter tube drives the magnet at the upper limit of signal strength, regardless of the position of the manual control. As in the RA-4 receiver, duplicate audio channels are provided for use by two pilots.

The switching of the receiver circuit is accomplished by a motor-driven control, which drives the rotary switches in the receiver. The eight frequencies may be arranged separately or in groups of two, with an auxiliary switch for selecting either of each pair.

The weight, including tubes and crystal units is 35 pounds. The power consumption is 1.6 amps at 12 volts and 45 milliamperes at 250 volts, the latter figure at maximum sensitivity.

Left, Bendix receiver unit mounted in Modified Lockheed Lockheed 14. Right, the new autotune loop, mounted in the unit. The loop and base drive weigh 8 lb. 3 oz., the loop amplifier and its control coil being 10½ lb. and weigh 10 lb. 4 oz. It is designed for use in the 100 to 415 kc. band.

# FATIGUELESS FLIGHT



Metals, as well as man, suffer from fatigue—from loss of strength due to repeated or alternating stress. But metals that are susceptible to this weakness have no place in aircraft engines.

Here are needed special metals, selected for their proven ability to resist fatigue, stress and wear.

To day the Nickel Alloy Steels are used

for numerous parts of these heavily burdened power plants because their superior mechanical properties minimize the danger of breakage and wear.

Through a partnership with Nickel the single steels are made unusually tough and strong, hence more enduring and reliable. Consultation on problems involving use of alloys containing Nickel is invited.

# NICKEL ALLOY STEELS

THE INTERNATIONAL NICKEL COMPANY, INC., 67 WALL ST., NEW YORK, N.Y.

AVIATION  
October, 1937







Today, Elgil engineers are cooperating in a continuous program of further bettering the airplane's power plant and propeller. The Moys of today are talking of speeds of 500 and 700 miles per hour, and that means still further improvement of fuels and engines. Elgil Gasoline Corporation, New York City.

# NEWS

DANIEL SAYRE  
C. E. McKenry, Profile Court  
Halsey Stables, Winchester  
E. E. Lewis, New York

DECEMBER 1937

**Beasts Maritime Commission Control: Says Normandies Not Tossing Fish**

Members of society's past in some transport had been favorable but limited, some ten days later Chairman Kennedy issued a special report on "Aircraft and the Merchant Marine" with special acknowledgment of the assistance to Greece and Turkey. Since it barely brought in one of the most complex situations in the history of the up-and-coming air transportation industry, this second report again cheerfully spun a series of tender sentimental tales. We present our abridged summary ending of the report's 22 pages:

"The general conclusion of this study is that the aircraft industry has been founded on a fundamentally realistic, probably slightly optimistic, construction."

... The elimination of intermediate landings in the Atlantic, whether in the one-way leg of Newfoundland or the return and low ceilings of the Azores, will reduce take-off and landing risks. Bulk flying boats will require the great advantage possessed two years by the de Havilland double-engine amphibians to overcome these obstacles.



Wah regard to comfort, the easiest of all transports to sit in is the airplane. . . . The berths in the flying boat designs are much wider and more comfortable than those in a Pullman. . . .

In order to study relative costs, it is essential to find a basis in a number grade costs system. The obvious one

[illegible]

The two days of the American Biologists including the Miller program features this year. The Dinner will be held at the Hotel Williams in New York City, Dec. 15. Following it, Prof. E. H. Reade, Jr. of Cambridge University will deliver the Wright Brothers Lecture, awarding the Wright Brothers Award, and the American Society of Mammalogy will present the Guggenheim Award, the Spence Medal Award and the Linnaean Society Award will be conferred. On Dec. 27, Prof. Jones will receive the Wright Lecture in Paleontology. The Annual Technical Meeting will be held on Dec. 28.

to choose at this time is the transportation of passengers across the Atlantic. In view of the fact that an American supersonic costing \$20,000,000 would take three years to construct, comparison is made to divisible and flying boat designs which are now under complete

(Continued on page 64)







### What's What in This Month's News



from his service as a pilot and instructor in the Air Corps, and his present commission as captain in the Air Corps Reserve. While studying law, commercial aviation was his avocation and in 1934 he was appointed Chief of the Enforcement Section of the Bureau of Air Commerce, later being detailed to the Inspector's Office of the Department of Commerce where he was assigned to the legal affairs of the Bureau. He came back to the Museum in April of this year as Chief of the Enrichments and Enforcement Division.



**CITIZEN:** COLONEL ARNOLD D. TUTTLE, from the Air Corps, after 35 years of Army service. For the past four years Colonel Tuttle has been Commandant of the School of Aviation Medicine at Randolph Field, and has supervised the training of flight surgeons for the Army and the Navy. He goes now with United Air Lines as medical director and chief flight surgeon. His duties will include instructing crew members in emergency procedures.

**APPOINTED:** J. J. O'Donnovan, as general traffic manager of Pennsylvania-General Airlines, succeeding W. J. Austin. In 1927 O'Donnovan joined Clifford Dill Company, first airline service established over the Alleghenies which was later absorbed into Pennsylvania-General Airlines. From 1928 to 1939 he served as Division Traffic Manager for the line at Washington, and in May of this year was transferred to Pittsburgh as assistant to President C. REHRIG, Mpls. 104 headquarters remain at Pittsburgh.

**LEWIS** DRIVERS D. BRYAN, MARION NASH, MARC, DONALD DODGE JR., GEORGE THOMPSON, and the 1957 Buick Wildcat. The racing couplets in the center reads of the TRA Transports, all to serve initially as First Officers. One look five second place at this year's National Air Race, and in 1959 won the N.A.A. "Champion Racing Pilot" in 1964 and 1965 he was the winner of the Freddie Lord Trophy for passenger airlines at Kansas. Thompson, co-lecturer in the design of his own racing planes, has held the Gyro and Henderson Trophies, was Thompson Trophy winner at Cleveland in 1955, and in the same year won N.A.A. "Champion Racing Pilot." Book established two racing records, in Illinois and back in 1951; from New York to Mexico City to Los Angeles in 1952, in 1953 and 1956 he is



THREE MURKETEERS FORWEAR THEIR RECKLESS WAY  
TWA's annual First Officers' Race. Capt. Ray, Harold Menden, Robert Earl



*Boissard, Euthymus consociatus* nov. consociatus. Euthymidae

anthropological research from the elm  
for the University of Pennsylvania

**ELECTED:** Major E. W. "Shorty" Schroeder, as vice-president of United Air Lines. For the past six months Schroeder has held an unpaid executive post with United (Aircraft, June, 1937), where his major interest has been safety in operations. He has been active in aeronautics for more than 25 years, and before joining United was Assistant Director of the Bureau of Air Commerce.

**ELECTIONS** Chairman, EUGENE C. DRENNAN, CARL HOFFER, FORTUNE W. KATZ, JR., as president, was present, and secretary-treasurer of the Air Transport Association of America. Elections took place at the annual meeting held at Hot Springs, Va. on Nov. 1 and 2. The new Board of Directors is composed of T. K. BRANNIFF, JACK FYFE, W. A. PATTERSON, E. V. RICHENTHORN, C. B. STONE, C. DENNIS MORGAN, and GARRETT PITTMAN. The two lost re-election CHAS. HUNTER and PAUL COLLIER.

**EMERGENCY:** Colonel CHARLES A. LINTBERRY, from his temporary retirement from American affairs. Colonel Lintberry accepted a five-year renewal of his commission in the Army Air Corps Reserve, and also attended a meeting of the Liberator Society in March in October. Mrs. Lintberry accompanied the Colonel on his flight from England to Malak.



STILL IN THE SADDLE  
A.T.A. member, Col. Barry

**1925.** LARRYSON BOWMAN, Wisconsin, 49-48, on November 6. Widely, a ship builder, was active in socialist agitation during the World War, and in 1915 lost the Fort Snelling magazine becoming president of Bowditch Marine Company. In 1921 he sought to purchase from the government the carriers Lexington and San Diego for conversion into passenger liners as well as airplane carriers.

... Count Ferdinand von Berthold, nephew of the airship inventor and his associate in the development of lighter-than-air craft. He died in Berlin on Oct. 28, at the age of 62.

## in Tabulariis Regibus

**AIRCRAFT MATERIALS AND PRODUCTS**,  
By George F. Tillerton; Pitman Pub-  
lishing Corp., New York 1957, 328  
pages, \$3.50.

Tutorian, an aeronautical engineer of extensive background who is now on the staff of the Grumman Aircraft Engineering Corp., has put together here a thorough and practical treatise on all modern aeronautical materials and the effect of various manufacturing processes on their properties. Recommended for inspectors, designers, purchasing agents and students.

AIR STRATEGY, by Lieutenant-General M. M. Gellman, Gale & Polden, Ltd., London 1986. 224 pages. 7s. 9d.

Am. Power and Assoc., by Wing Commander J. C. Stewar, Dafford University Press, London 1929, 215 pages, 3s.

Alphonse "Wunt" backs his argument with a realistic conception of aerial warfare as it will be fought if, when, and on two front class air powers meet in a crisis. Both books deal primarily with Germany's most disastrophous military defeat, but the author's credibility in this regard is unquestioned. The first, written by an officer formerly of the Russian Army, deals with human aspects of general air warfare including the now-famous advocacy of "Jug-Burntums" by General Guderian. The second, written by a specific tactical problems raised by modern trends in military and aerospace. While Commander Shores' outline of the second volume, has been a lecture on his subject for some years at the RAF Staff College, it is not a book that should be read for its potential role of aircraft in non-military operations with ground forces.

**Synthetic and Adrenergic Mammals.** by Herbert Robert Foye, B.S., D. McGraw-Hill Book Co., New York, 1957. Pp. xviii, 226.

For 6 years all our line operators personnel and the vast majority of purchasing automobilists have been asking for "one good book on car maintenance and its associated applications." This is Dr. Devera, professor of Dr. Roszak's graduate school, a M.I.T. and at present in charge of U. S. Weather Bureau. Air Conditioning Section, was absolutely winning for a task which he had undertaken with very limited success. The book is now for members of the Junior First year. But how those who have a good background in physics and a grasp of elementary aerodynamics it is the one for first.

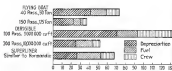
## LOADING BOOSTS CLIPPERS

(Continued from page 21)

technical solution and are easily to construct when an order is placed. In addition, consideration is given for the future in design of 25,000 cu ft. carrying 300 passengers, and 350,000 pound thrust loads carrying 350 passengers. Both of these are considered by increased weight to be actively feasible. The time of cost in which this study is based are: (1) Depreciation, (2) Fuel consumption, (3) Wages and salaries of operating personnel.

Full capacity at seaward. No consideration is given to landward passenger as the theory that transatlantic mail can be provided entirely by cable telegraph. The experience was 1,000 transatlantic hours per passenger crossing and the flying boat also only about one-fifth as much—1,000 hours per passenger crossing. The daylight loss 1,000. The lower installed power . . . and the small number of passengers carried is more than compensated by the brevity of a flying boat crossing. . . . When it comes to still further increase in speed, the flying boat has by no means attained its limit. The experience, even with its power doubled, could increase its speed only a few knots. There is considered here only the minimum which is immediately practicable: (1) One experience along one mile per week; (2) Two distances making 21 day crossings on seasonal schedule of 2 to 4 per week; and (3) 4 flying boats making daily crossings each way. . . .

The American construction cost of the airplane is estimated to be 200,000.00. The construction cost of an equivalent passenger capacity in dirigible would be about the same. The cost of equivalent passenger capacity in flying boats is estimated will be \$10,000,000. The experience is dependent on a 30-year basis, the dirigible on a 5-year basis, and the flying boat on a 3-year basis. But present designs and an 8-year basis for flying boats are very much less on the flying boat than on the experience of the dirigible.



LOADING'S PRIZE EXHIBIT. A comparison of loads cost (data for real transatlantic transport forms).

slightly because of the fact that, on a one-night passage, less service is required for the passenger comfort than would be the case for a longer period (the short for even 6 months).

The volume of transatlantic traffic must be approximately 5,000 to 5,500 pounds per business day, and the average number of first-class express passengers paying somewhere near the proposed airplane fare (estimated at \$100 a passage) is likely thousands a year. The suggested service of a daily airplane . . . with a 40-passenger capacity rate, on a basis of a 50 per cent load factor, will have a capacity of well over 5,000 pounds of mail . . . could carry very nearly all of the daily first class mail. The five per cent rule to Europe . . . would amount to 25 a pound . . . (therefore) 5,000 pounds of mail would bring an added income of \$25,000 per day to the airlines if they were paid an amount equal to the full postage.

It would appear, therefore, that three aircraft (two-day to Europe by airplane and 11 days by dirigible) may, in the near future, be operated at a profit, and with a three-day or possibly less than that of a superliner. Such full service, with much capacity for a large part of the passenger mail and express traffic will mean appreciable service to last much of its expected life, justification for a large class of traffic.

(If the shipping companies are not to add service to their fleets, they will undoubtedly find considerable trouble in independent airline companies.)

### Dirigibles Still of Use

The dirigible service still has a place in the merchant routes field. As may be seen from the cost comparison already set forth, it may become superior to the experience of large passenger flying boats has made the past experience of the dirigible for passenger service much less important. The dirigible is much more expensive in first cost, and its requirements of large hangars, mooring masts, balloon cables, and large landing crews or landing machinery, are certain to make its operating costs higher per passenger

than that of the airplane. For approximately equivalent loads dirigible still have much larger range than flying boats. Thus, the increase in range of the flying boat will enable it to negotiate the Atlantic to Europe with as much reserve in range as the dirigible formerly had. But there are some very long non-stop routes . . . such as San Francisco to Japan, San Francisco to Australia, New York to Asia, New York to South Africa, etc., which are at present still too long for successful operation of flying boats of under 150,000 pounds without refueling or intermediate stops. The dirigible will be particularly for this long range work.

### Landships Considered

The land type of airplane with four or more engines is capable of effectively spanning transatlantic distances. Yet it is distinctly limited in comparison with the flying boat by its take-off requirements. There are no landing fields now available large enough to take care of the heavy wing loadings that can now be born of the water. A good deal of work is being done on the "landship" aircraft, that is, by catapult, launching, etc., etc.

But, as long as the flying boat, suitably approached, the landship in load capacity and in high speed, there does not appear to be any valid argument for abandoning the flying boat. Its added safety over water as well as its ability to land in many unprepared harbors at any time, its unsuitability as a substitute over the landships which could at most float for merely a few hours if forced down at sea.

### These International Rights

Up to the present time the negotiations for air rights in overseas countries have been made almost entirely by one American company (Pan American, of course) for its own overseas movements. Planning this field, it acquired exclusive rights for its own operations in the former trade routes of the world and to her present, therefore, in a unique position in this respect. . . . The control by the British government of Bermuda and Newfoundland has led the American company into various arrangements and agreements with the British Government. Airways designed for their mutual benefit, and finally to the issuance of permits for the operation by these lines by the Department of Commerce. A British permit dated 23rd of January, 1937, signed by the Director General of Civil Aviation . . . grants to Pan American Airways the right to operate two trips a week between the United Kingdom's military airport serving London and the civilian port serving New York City—no condition that a similar permit of approval be granted to Imperial Airways by the United States Government. The Department of Commerce permit is similar except that Pan American Air-

# New and Faster Wings for National Defense



## 210 Curtiss P-36 ARMY PURSUIT PLANES

Curtiss is proud to have been selected by the U.S. Army Air Corps to build 210 P-36A Pursuit Airplanes.

These swift and powerful weapons for National Defense will give the U.S. Army Air Corps fighting planes that are second to none in the world. Suitable planes for pilots recognized as the most efficient in the world.

Curtiss was awarded this contract for 210 Pursuit Airplanes—the largest peace-time order ever placed by the U.S. Army Air Corps—after keen competitive tests ranging over a period of many months.

Curtiss takes this opportunity to pay tribute to the U.S. Army Air Corps, who cooperated in the design of the P-36A Pursuits. Airplanes of such advanced design and outstanding performance have given the Air Forces of the United States world leadership in the air.

CURTISS AEROPLANE DIVISION • CURTISS-WRIGHT CORPORATION  
FLYING "The Powers of Aviation" NEW YORK



**Curtiss** THE LEADING MANUFACTURER OF PURSUIT AIRCRAFT IN THE UNITED STATES





## AS OTHERS FLY IT

A Bird's-eye-View of Aviation Abroad



**VEPRICH RUSSO.** A zeppelin stayed up 118 hours in 1933 waiting for a Russian revolution to arrive (that was a world record). Now the Russian navy is developing airships under General Yakovlev. While there's a new one. Their new ship, the V-1, recently stayed up 118 hours in a cruise over European Russia. (Veprich Russia = Up Ship.)

**RLM has stepped up frequency** on its Aerostandarten service from two to three round trips per week. The German government has established a weather station near the top of the Jungfrau which stands up 15,078 ft. above sea level. Helms Zeppelin has flown the Focke-Wulf windmill plane on a 47-mile run-up from breaking the existing world record by 11 miles. J. van Buren, who got out with No. 1 Dutch airship, has set a new all-out record between Australia and England—3 days, 16 hrs. and 35 min. in the remote case that you're interested.

Air France has set the first of fourteen new 300-passenger passenger transports to be delivered and service between Paris and Marseilles.

The British expect to have a new stretch-curve, R. M. & A. & A. (London) in April, or, possibly by next July. These were British carriers are already under construction. The last for a fifth was to be last down last month.

men from the Netherlands that the shoreward Dutch Zeppelin line has been that country and the East Indies has become only a remote possibility. Such a service, for one thing, would not be eligible for Nelson prohibition under the current U. S. act.

**Last recent get-together of Lima**, held as the First Inter-American Conference on Aviation and in which the U. S. sent a full delegation and



**ROADS TO EDEN:** The steadily expanding web of international airways in Chile has been set to build the prestige and speed the communication of the Second Spanish Empire—Spain's airways are still in the air. Latest addition is the new privately owned service to Bogota, Venezuela and Buenos Aires.

the carrier Banger, didn't get together so very much. Some very pointed progress toward a two-continent weather network was effected. But only limits of the Treaty of Amity, non-political and representative. And of those Argentina and Brazil stood out against a two American air only on the grounds that none of the American countries save the United States had even marked out under the same laws. The conference's action setting up an Inter-American Permanent Commission on Aviation to draft such a code seems therefore rather futile. Brazil and Argentina who refused to join an Inter-American Federation in Civil Aeronautics on the grounds they already belonged to the I.A.T.

**Airships for military spotting**, under study in America here the standard official British interest. The Air Ministry has recently ordered five "jump-off" units (the first with land and naval forces). Powered with 350 hp. Bentley Grant engines, the new ships are to carry two side-by-side will have motorized gun turrets.

**Spinning of John Bull Imperial Airways** has been taking steps toward extending its 45 service operations out of London. Egypt. The Empire Flying Boat Company is making survey flights between Alexandria and Koweik. Three QANTAS planes from Australia have finished flying boat service in South America. Four more are expected for further introduction. The Air Ministry has raised the limit for flying club subsidies from \$125,000 to \$175,000. Club planes must be 181 August 2 against 184 Jan. 2, 1934. G. E. Gaudin's speed of 237 m.p.h. with which he won the Knight Cup Race for the second year in succession, set the British industry going another—so well it might, since it was made over a 1,000 mile course with more knots in it than a pistol.



Official Photograph U. S. Navy

# GRUMMAN F3F-2

## Fastest U. S. Navy Fighters, powered by WRIGHT CYCLONES

Wright Cyclones will power the 81 Grumman F3F-2 fighters to be constructed for the U. S. Navy at the modern, new factory of the Grumman Aircraft Engineering Corporation at Bethpage, Long Island.

The Grumman F3F-2 is the latest type fighting plane now in production for the United States Navy. It is superior in speed and service

coiling to any U. S. Navy fighter now in service.

Again Wright Cyclones lead the field! Engines of this type power many of the world's most advanced types of military and commercial aircraft—ranging from such powerful fighting planes as the Grumman Navy F3F-2 to the twin-engine Douglas Army Bombers and four-engine Boeing Bombers—the formidable "Flying Fortresses."

"Fly With Wright the World Over"



**WRIGHT**  
AERONAUTICAL CORPORATION



# Operators' Corner

An exchange of ideas on the problems of the commercial aviation industry

**QUESTION 47:** What has been your expert opinion on airline expansion in airports that have expanded the terminal? Have you found the ratings for the private pilots' perspective on other airlines? What would have you said to convince the airlines to expand their facilities?

## Read the Policy

Since we are concerned here with completely failed in their plans for the purchase of airplanes by the restrictions placed upon them in their life insurance policies. Others have found after making and interpreting their policies that the restrictions were not as strict as they had supposed and that they would be permitted to own and operate their own planes without violating their insurance.

An insurance man informed me that an individual who did not have the restriction of flying as the life insurance contract would not have to wait for the incontestable period to expire before starting his flying activities. He went on to say that a would be necessary for the insurance company to prove that he had had this restriction in order to void the policy.

Some insurance men take a liberal view of flying and even in some forward to the day when insurance policies will allow greater freedom for those who risk to fly. Others whose usual avocations are confined within the safety confines of their experience today won't even take time to be shown what is going on in aviation.

They turned me off a venerable old gentleman who lives a couple of miles from my home. He has never "lost his" in an automobile and says he never will. I hope his pigheadedness will be defeated when he takes his last ride in a gasoline powered machine. His death in some time to come we shall look with similar wonder upon those hangovers of a bygone day who neither will look in an airplane themselves nor permitted others to do so.—**CLARENCE B. BRANSON, President, Midwest Air Service, Highland Park, Ill.**

## Liability Coverage Helps

It is not only the insurance that the present high cost charged by life insurance companies on policies issued to pilots, and existing policies which carry riders limiting the coverage in the event that the policy holder should take up aviation, but in many cases presented surplus value. Such insurance created by this attitude of life insurance companies has been especially responsible in cases where the rate would have involved an expensive airplane, in all probability fully equipped in a manner comparable with transport airplanes operated by scheduled airlines. My have in yet found no method of averting the difficulty resulting from policies which carry riders.

In the case of private pilots, who already possess the license and who are contemplating taking on new policies

or increasing the coverage over old policies, we are, I believe, advising them to wait until such time as some life insurance company will take the initiative and write life insurance for them at a low reasonable figure. A company taking this initiative would remove our whole burden.

We have found a possibility of covering the life insurance difficulty in our charter service by providing our passengers with passenger liability insurance policies to reimburse them in case their own life insurance would not apply.—**W. W. WILSON, President, E. D. Wiggins Airways, Inc., Newstead, Mass.**

## Discourages Flying

THE EXTRA COST of life insurance when a substantial amount is carried by a private pilot is a great handicap and I know of several of my own friends that gave up flying and sold their ships because of it.

I could not find any way to reimburse them that they should pay this extra charge which is the cost of a dollar but has \$500.00 per year above its actual insurance rate. The remedy is in the hands of the insurance companies and should get their consideration.—**JOHN W. MILES, Executive School of Aviation, Kensington, Pa.**

## Three Sales Lost

I SAID IN MY LAST LETTER that the airlines are not able to pay cash for the instruction and plane, if the insurance factor had not been a deterrent.—**W. D. CARR, General Flight Instructor, Memphis Airport, Pittsburgh, Pa.**

## Next Month's Questions

**QUESTION 48:** What specific methods have you used to promote your business? What have you achieved? Have you been able to secure the best type of service and to whom have you secured the best type of service? Have you been able to secure the best type of service and to whom have you secured the best type of service?

**QUESTION 49:** What are your methods in the new Civil Air Regulations? What are the best type in the new Civil Air Regulations? What are the best type in the new Civil Air Regulations? What are the best type in the new Civil Air Regulations?



Only commercial airplane now in the Milwaukee Zeppelin operated by Anthony G. Jones



STANDARD RACK MOUNTING of equipment—Early version shown. Northwest about 1940.

CENTRALIZED RADIO CONTROLS are featured—General version shown—Simplest installation.

RELIABLE VACUUM TUBE MOUNTING—General version shown—Simplest installation.

**NEW SKY ZEPHYRUS SCORE A FIRST!**

**Bendix Radio Equipment Used In First Commercial Radio Installation Engineered Simultaneously with the Airplane**

When specifying accessories for this L-14 Super Constellation, Northwest Airlines selected Bendix Radio equipment. From the last line a commercial radio installation has been engineered simultaneously with the airplane. Bendix engineers, in cooperation with the Northwest and Lockheed organizations, assumed complete responsibility for every feature of the equipment. Control panel and placement, location, layout, wiring and a host of other details all engineered by Bendix. Their refinements include volume control, signal level meters and emergency intercom system and a host of other details all engineered by coordinated engineering.

The radio unit featured on the airplane engineer and research facilities of Bendix. Bendix and its technicians are present—and they are present by the excellence of workmanship which characterizes all Bendix equipment. The Type RA-2 eight channel 100 watt aircraft radio transmitter equipped with electrically operated channel selection is a typical example. So is the Type RA-2 eight channel aircraft communication receiver or the RA-4 Radio Range Receiver. Consequently equipped, and incorporating the new "change in radio" for details of complete engineering service on any type of radio installation or maintenance on specific equipment, officers require less.

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**BENDIX RADIO CORPORATION**



## COMMUNICATIONS TRANSMITTER

The communications transmitter of the future made possible TODAY by the engineering skill and experience of BENDIX designers. To operating officials it affords greater operating and maintenance economies, low line charges for remote control and lowest costs on necessary channel changes. It also means a new high in reserve power, in simplified speedy operation, noiseless switching and accurate automatic tuning. Penetrating electrical assembly and mechanical construction provide a maximum of subsidiary service flexibility and an operating economy amplified by lowest obsolescence per dollar cost.

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FREQUENCY RANGE . . . 1,500 to 30,000 Kcs  
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CHANNEL SWITCH — simplified, rapid, quiet  
DIAL OPERATION . . . over single telephone pad controls  
all channel selection and extreme switching functions  
HIGH SPEECH FIDELITY AND LOW HARMONIC CONTENT  
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CABINET — sturdy, accessible, modern.

BENDIX RADIO

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CHICAGO, U.S.A.

## "Here Goes Nothing!"

(Continued from page 25)

ings in the early standard stage of system adjustment. The spin-down saves the airplane from loss and frequently saves the life of the test pilot—stepping from a spinning airplane has proved fatal to some of the world's best test pilots. The spin characteristics are customarily explored by first making stalls, then half-turn spins, gradually working up to steady spins with various portions of the control surfaces as the recovery characteristics become more and more constraining. Such methods of spin-testing where followed effected a reduction in test airplanes and killed pilots. It happens quite frequently that an airplane thought to be perfect in spin-recovery and after many apparently perfect spins suddenly develops an uncontrollable spin without warning. Possibly a change in wing characteristics or some other slight alteration may have made the difference in spin attitude. Then the pilot, unprepared with a spin-chart, may be killed, either through struggling too late with the controls or through jumping and being hit by the spinning airplane.

Spinning is dead ends in a new manner in the third stage of test piloting. The test pilot will, in the first place, have a considerable voice in the design themselves, and will have speedy the size and disposition of control surfaces which from his knowledge of spinning, he knows will give satisfactory spin-recovery as well as the most excellent controllability. A spin-recovery model is then made of the airplane, correct in every detail of center-of-gravity and moments-of-inertia, and this model is tested in a low-speed tunnel to determine exactly its spin-recovery characteristics. Usually where the airplane has been built in accordance with the requirements that no possibility of failure yet by at this point considered, it is flown empirically and spun with an adequate spin-chart as described above.

Not only in emergencies like spinning but in all other phases of flight testing can we trust the three stages to flight-test approach. Nowhere is clarity in the determination of controllability in the designs made evident. Although a mistake in designating effective or measurable controls may and sometimes does, result in a

crash on the first flight, the determination of safety at this point is frequently left almost completely in chance. If a rudder is overbalanced the test pilot of the first stage finds it out after he is in the ground, possibly even at an altitude of flight from which recovery is exceedingly difficult.

In the third stage controllability is approached from an entirely different angle. The effectiveness of controls is checked and established by several aerodynamic methods in preliminary design. Controlled stability computations determine quite closely the quantitative values in turning, pitching and banking. On large airplanes the time required to execute a turn, bank or pitch can be computed and is now frequently specified by engineers for their new air loads. The control forces or hinge moments are more accurately of exact comparison, but wind tunnel testing of control forces is now becoming common and will probably eventually be done on all new designs departing appreciably from standardized practice or pattern. It is only in this way that we can reduce the unknowns of testing airplanes to any such minimum as the few which remain. For new stresses shifting down their wings for a trail run at sea. Yet it is not infrequently seen now that large aircraft companies, which are still in the transitional attitude between the first and third stages, run great risks on test flights of new aircraft by testing controllability up to the chance that the test pilot will be able to get the ship down in case the controls have not been properly oriented as to aerodynamic overbalance, underbalance, or effectiveness.

Another distribution of changing attitudes in systems can be seen in a review of the present state of flutter. Control-surface flutter is a phenomenon still imperfectly understood by many designers. Years ago when it occurred it was followed by the loss of the airplane through disintegration under the forces of vibration and aerodynamic loads. Even if the pilot was not killed he could usually supply little information after such an occurrence as to the sequence of events because of the rapidly with which failure followed the first appearance of flutter. Later, when the success

of flutter yielded to investigation and theory and tests, there were still many pilots as well as designers who balked their heads in the wind, refusing to believe that dynamic balancing might be a very useful solution. For something there are already using "third stage" designers today who will build an airplane without a complete dynamic analysis of the control surfaces and suitable provision for flutter prevention.

A honeycomb device built for long distance saving was supplied by the test pilot of a tendency to over-rotate although he was unable to clean any gross bending or excessive vibration. The designer refused to believe in flutter danger or to provide flutter-preventing mass-balance. During tests on a third ship involved in severe danger and structural failure of the stabilizer, vibration static balance weights were added and found to provide freedom from danger at the speed at which it had occurred before. Whether this proved that the speed or resistance had been picked up or down by the addition of the static balance weights was not known, and only complex dynamic balance weight laws safeguarded all speeds. "It was known that there are 'flutter islands' in the speed range outside of which flutter would not occur at any speed and in which it might occur only if an interfering device were present at the correct frequency to set the structural stiffness and moments of inertia down. The designer had not realized a slow dynamic balance because the weights would, on this design, have to be placed outside the surface of the airplane, the test pilot threw in test the flutter possibilities by shaking the stick to drive at all speeds in order to supply the interfering factor which might eventually be applied later by a bump in the pressure field at that rate." The designer felt that this was all much too complicated and uncertain. Later the rotor fastened to glass in a case and the pilot escaped only by pushing forward with his joystick. A change in attitude of the designer would have saved both the bomber and the stabilizer at the former case which failed in tests. The three stages of testing airplanes are clearly shown in these instances. They typify the progression observable in the behavior of all living beings from the trial and error methods of solving problems toward the eventual realization of complete scientific solution in terms of knowledge and planning.

## Celestial Navigation

(Continued from page 25)

the azimuth. The computed altitude is then compared with the measured sextant altitude to determine the position of the observer relative to the assumed position.

**Circle of equal altitude**—Since the altitude of a heavenly body would be the same for every position on the earth's surface equidistant from the body's geographical position, it follows that one altitude will not give a definite fix; nevertheless it will give a circle on which the observer's position must be.

In Fig. 2, S is the geographical position of a star at a given time. If the observer were at S at the instant, he would see the star in his zenith; i.e., the observed altitude would be 90°. But if he were, say, 2,700 nautical miles away from S, his altitude would be over some point S' on the altitude circle instead of at S. Now we know that 1 nautical mile represents an angle of 1 minute of arc, so 2,700 nautical miles represent an angle of 45°. Then it is clear that the star S would be 45° away from the observer's new zenith. Tracks would now indicate that there were a great number of positions on the earth from all of which this heavenly body would have the same zenith distance of 45° and all these positions would lie on a small circle bearing the geographical position of the star as its center. Such a circle is known as a circle of equal altitude. Having measured the altitude of a star and knowing its geographical position, the navigator can find the circle of equal altitude upon whose circumference he must be situated. By immediately obtaining a second star, he can determine a second circle of equal altitude, which also passes through his position. The intersection of the two circles of equal altitude does his position. While in general there are two intersections of these circles, the navigator always knows his position closely enough to eliminate one of them.

The line of position (L) is any used to indicate the whole earth or any one part, only that part of immediate interest in the navigation is shown. Hence the whole circumference of circles of equal altitude would not appear, as the radius of the circle is usually large, the small, slightly curved arc in which the navigator is interested may be replaced by a



Fig. 4 How to obtain the line of position

straight line without appreciable error. This line is called a line of position or position line or Sumner line after Captain Thomas H. Sumner, an American navigator, who discovered it in 1822. The direction of the line of position on the map must always be at right angles to the radius joining the intersection point to the geographical position of the heavenly body; in other words, at right angles to the true bearing or azimuth of the heavenly body at the instant at which the observation was made. All the methods of interest to the navigator are based on the method of Marco St. Hilaire, which makes use of the difference between the observed altitude and the altitude computed with the aid of an assumed position.

**Finding a position line on a chart**—To lay down a line of position on a chart by this method of Marco St. Hilaire, it is necessary to take the dead reckoning position, or some remaining point close to it, and to compute for this assumed position the altitude and azimuth of the observed body at the time at which the observation was made. If the altitude observed with the sextant agrees exactly with the computed altitude, the altitude difference is zero. Then it is only necessary to draw from the assumed position A (Fig. 3) a line of bearing AB along the computed azimuth of the body and a line CAD at right angles to this; then CAD is the required line of position.

If, however, the observed altitude is greater than the computed altitude, say by 15', which equals 15 nautical miles on the earth's surface, the observer is nearer to the geographical position of the heavenly body than the assumed dead reckoning position would indicate. In this case the position line must be moved closer to the geographical position of the heavenly body by the amount of the altitude difference. Hence for intercept AB, it is laid off 15 nautical miles towards the body along the computed azimuth, and the position line PBO is drawn, at right angles to AB.

The 40-20 two position line method is used in Fig. 3 are found, the actual position or fix, assuming there are no errors in the work, will be at their intersection. If three position lines, such as AB, CD and EF in Fig. 4, are determined, the true position may be reduced to an intersection within the shaded triangle or "cocked hat." To determine a number of lines of position, it is necessary to take early simultaneous observations of different heavenly bodies or to get radio bearings of stations of known position.



Fig. 3 Finding altitude difference and position

It should be remembered that the center point of position are at right angles to each other, the closer and more observed their intersection will be. As the angle between two lines of position diminishes the intersection becomes less clearly defined, and, at 15° or less, little faith can be placed in the fix obtained.

Whenever stars are visible it is an easy matter to select a pair of stars whose position lines will give a good "fix." During the daytime, however, the sea is usually the only heavenly body available. To marine navigation it is the practice to derive a position line from two observations of the sun several hours apart by moving the required line of position.



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# Universal Preference Pioneer Instruments



## Flight Panel



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## Celestial Navigation

(Continued from page 20)

the first line parallel to itself for a distance equal to the estimated error between the observations, it is made to cut the second line, and that gives a fix.

In the air, however, this is of little value because the high speed of the aircraft, and possibly of the wind, makes it impossible to estimate accurately the true bearing observations. It will usually be found more satisfactory to consider these lines separately in celestial navigation. A single position line will always always give valuable information and often all that the navigator requires.

Consider the case where the position line is parallel to the coast of the aircraft. Then the distance from the coast, as laid down by dead reckoning, will give the navigator an indication of the accuracy of his dead reckoning and perhaps warn him of a change in the direction or intensity of the wind. Similarly, if the position line is perpendicular to the coast, the navigator can determine the approximate distance he has traveled. In this case the course plotted by dead reckoning may be regarded as a position line, and its intersection with the astronomical position line will be the most probable position of the ship.

It should be remembered that at certain times the Moon, Venus, and Jupiter may be available in the daytime, and may be used to give additional position lines that will cut the Sun line, or cut each other.

**Position lines by sunsets.**—The careful reader may wonder why the sun gives data not just as fix by using the observed altitude of a heavenly body in conjunction with its observed azimuth. The answer is that he can fix it, but it is not worth while because the accuracy obtainable is very poor.

The altitude can be measured with sufficient accuracy, but the azimuth cannot be obtained with a comparable degree of accuracy. On an average the azimuth determination of a body at altitudes up to or about 30° is accurate to about 1°, this would give a range of information with a position line at right angles to the sun with a 30 to 60 miles each way.

The accuracy arises from the great distance of the geographical position of the body from the observer's position. As soon as this

distance is decreased by using an other body having a nearer geographical position, the altitude of the body is increased so that, although the distance is thus small enough to reduce the error due to distance to a reasonable amount, the altitude is so great that the observation of azimuth becomes very inaccurate owing to the effect of errors of the level of observing instrument.

**Limits of Accuracy.**—To be of great use to the navigator, the results obtained by celestial navigation must fall within the limits of speed and accuracy required in the air. We may approach the problem by saying that if a position within five miles may be obtained in five minutes or less, then celestial navigation is of real value to aviation.

Further experience may change these figures somewhat, but the principle is correct, namely, that methods that are in paper useful in navigation must be within certain limits of speed and accuracy. Actually, a line of position may now be worked on the air or less than three minutes during the daytime, if working with stars during the night, a definite fix may be determined in about one minute, with an average accuracy of about five miles.

**Summary.**—Briefly, celestial navigation consists in finding and plotting on a chart one or more lines of position. One line of position gives the navigator certain useful information. Two or more lines of position determine by their intersection a definite position or fix. To determine a line of position, the navigator must take the following general steps:

- (1) Observe the exact altitude of a known heavenly body.
- (2) Note the exact time of the above observation.
- (3) Take the declination and the Greenwich hour angle of the observed body from the Almanac for the Greenwich Civil Time of observation.
- (4) Apply the assumed longitude in the Greenwich hour angle to find the local hour angle.
- (5) Compute the azimuth, using an assumed position.
- (6) Compute the azimuth, using the same assumed position.
- (7) Compare the computed and observed altitudes to obtain the altitude difference.
- (8) Set off the altitude difference along the course or assumed and draw the line of position at right angles to it.

The practical application of the principles outlined here will be discussed in the succeeding installment.

## Waco N

(Continued from page 30)

drum brakes on the rear wheels are a part of the left mainder pedals. They are toe-operated. Retractable under pedals are provided for the right foot rest and a three-over wheel makes the airplane easily flyable from either seat.

Brakes actuate the rear wheels only and are used for steering on the ground, the front wheel being of the caster type and non-steerable. It is possible to describe circles around either rear wheel up to a pivot.

A parking brake is actuated by a lever immediately to the right of the control column accessible to either of the front seat occupants.

Lanterns are mounted in a shock-proof indirectly lighted panel with ground wood finish which is duplicated in window moldings and other exterior metal trim.

A telescoping tab on one elevator is slotted in flight from a control wheel on the control column column. Flaps are controlled with a crank valve in the seat between the two front seats. A quarter turn is used from the full-on to the off position.

**Waco Model N characteristics:**  
Make and model of engine: Jacobs E-3  
Rated power at 8,000 ft.: 282-3,000  
Span overall: 34 ft. 3 in.  
Length overall: 29 ft. 8 in.  
Height overall: 9 ft. 7 in.  
Wing area: 276 sq. ft.  
Cruising speed: 145 mph.  
Empty weight Standard: 2,380 lb.  
Disposable load Standard: 1,276 lb.  
Pay load Normal: 642 lb.  
Wing loading: 43 lb./sq. ft.  
Power loading: 12.81 hp./sq. ft.  
Fuel capacity Normal: 20 gals.  
Fuel capacity Maximum: 55 gals.  
Oil capacity Normal: 5 gals.  
Oil capacity Maximum: 8 gals.  
Standard Empty weight: 2,380 lb.  
Top speed S.L.: 105% R.P.M.

154 M.P.H.  
Cruising speed optimum altitude (3,000-8,000 ft.): 144 M.P.H.  
Cruising speed sea level: 135 M.P.H.  
Landing speed sea level: 58 M.P.H.  
Climb first minute: 880 ft.  
Service ceiling: 14,000 ft.  
Fuel consumption at best cruise: 17 gals./hr.  
Range at best cruising speed with maximum fuel and 10% reserve: 622 miles.  
Landing gear wheel: 18 in.

1



Non-oxidized photographs show the smooth, bright surface appearance of the modern "Non-Oxidized" tube in contrast to the black oxide finish formerly used. (Mark end covered.)



"Ohio Special Non-Oxidized" Aircraft Tubing is now Standard in the round and available in section as well as special shapes.

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### Space Structures

(Continued from page 78)

after finding the first apex M, all we need is to project it to the line AP and we have a second apex. The two are sufficient to determine the locus of all the points M and we can draw this line through M' and the last point. As before, the intersection of this line with the line through S gives the projection of the true stress in the member AC, and the other stresses follow.

In view of the great variety of trusses and of the difficulty in avoiding mistakes in the work, it has been found necessary to create a nomenclature which helps in the procedure.

Fig. 50 gives the names of the various lines and points. The line AP is drawn through the tail end of the lead vector, hence it is called "Tail line" (T.L.). It is parallel to AL, which then becomes the "Tail member" (T). The line drawn through the front end of the lead vector is the "Front line" (F.L.) and member AC is the "Front member" (F). The line through P is the "Middle Line" (M.L.) and AD is the "Middle member" (M). The intersection of the M.L. with the F.L. is the "Auxiliary point" (A.P.) is

the same as the line through P is the "Auxiliary middle line" (A.M.L.) and the projection of AP on it is the apex of the stress member which is called the "Maximum apex" (M.A.). The projection of O on the T.L. is the "true width apex" (T.A.). The line ZA, MA, intersects T.L. at the "True point" (T.P.); the projection of T.P. to the F.L. is the top view in T.P'. The true M.L. drawn through T.P. gives the point R. The projection of the latter on T.L. gives R'. The line R'T.P' is the true middle line and must be parallel to the M member.

With the aid of this terminology the procedure may be described as follows:

1. Draw the two axes of the truss and of the load. Mark the members on the top view arbitrarily T, M and F. Mark the corresponding lines born in the other view the same way.
2. Through the tail end of the lead vector draw T.L. parallel to T and through the front end of the vector draw F.L. parallel to F. Do this in both views, as the top view mark Q, the intersection of F.L. and T.L.
3. On T.L. in the top view choose a point P and through it draw M.L.,

such, as intersection with F.L. as "A.P."

4. Project P on T.L. and mark P'. Draw A.M.L. through P' parallel to M in the side view, project A.P. on A.M.L. and mark point "M.A."

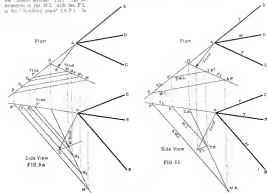
5. Project O on T.L. and mark "T.A."

6. Connect T.A. and M.A. and mark the intersection of this line with F.L. at T.P.

7. Draw M.L. through T.P. and mark point R. Project R on T.L. and get R'. Connect R' and T.P'. This line must be parallel to M.

The resulting stresses can now be laid off on the corresponding members and the lines must be so drawn to find the true lengths before making them. Notice that F, F', R, R', O and T.A. are T.L. points. T.P', T.P' and A.P. are F.L. points. M.A. is an A.M.L. point.

Such is the theory of the graphical method for solving space trusses. A systematic procedure has been worked out to make its application simple and mechanical and to point anyone to check its correctness. Details will follow in a second article.



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*Left:* Eclipse Propeller Anti-Ice Pump (variable output—12 or 24 volt operation) for application of ice removing fluids to propeller hub splitters and windshield.



*Left:* Eclipse Remote Control Rheostat (shielded) for propeller "Anti-Ice" pump.



*Above:* Eclipse Electric Motor Driven Ten Port De-Icer Distributing Valve for 12 or 24 volt operation and for operation of wing and tail surface Goodrich De-Icers.

*Below:* Eclipse Electric Motor Driven De-Icer Distributing Valve (see integral control valve) for operation wing and tail surface Goodrich De-Icers.

